



Jane's
POCKET GUIDE

MODERN MILITARY HELICOPTERS

TIM RIPLEY

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HELICOPTERS

T I M R I P L E Y



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Introduction

"Death from Above" was the famous insignia on the nose of Lieutenant Colonel Kilgore's UH-1 Huey gunship in Francis Ford Coppola's Vietnam War epic *Apocalypse Now*. The 25-minute long section of the movie where the 1/9th 'Air Cav' take a Vietcong-held village to the sound of Wagner's *Ride of the Valkyries* captured perfectly the essence of going to war by helicopter.

Since the Vietnam War the helicopter has been an integral part of every armed force, and rotary-winged aircraft have seen action in every major conflict and many small wars. This rapid acceptance of helicopters into the mainstream of military organisations in attack, reconnaissance, liaison, transport, medical and maritime roles has led one commentator to term them 'rotary-winged fighting vehicles'.

However, when helicopters first saw action during the Korean War, they were used by US armed forces for casualty evacuation and VIP transport only. Indeed, it was left to the French to first demonstrate the combat potential of the armed helicopter during their colonial conflict in Algeria. The 1950s and 60s also coincided with revolutionary developments in helicopter design, such as the tandem rotor and turboshaft powerplants.

The success of US Army and Marine Corps gunships in Vietnam spurred the Soviet, British, French, Italian, Israeli, German and numerous other armed forces to field their own fleets of helicopters for anti-tank and assault work. By the mid-1970s most armies had began programmes to procure specialist attack helicopters, leading to the development of the

current generation of Cobras, Apaches, Tigers, A 129s, Mi-24s, Ki-50s and Rooivalks. It must be duly noted that these developments were often made in the face of stiff opposition from air force 'blue suiters', who saw the armed helicopter as a direct rival in their own fleets of fixed-wing close air support aircraft.

No such argument was put forward by the navies of the world, however, as they had been quick to embrace the armed helicopter for the anti-submarine and anti-surface vessel roles, as well as more conventional air transport duties.

Indeed, the 1982 Falklands War proved the worth of the helicopter in naval warfare, protecting the British fleet from Argentine submarine attack, decoying Exocet missiles with electronic jamming devices and sinking enemy shipping with guided missiles. Five years later US Army and Navy helicopters provided vital protection against Iranian fast patrol boats in Persian Gulf.

The 1991 Gulf War saw helicopters employed successfully in a wide range of roles by Coalition forces, whilst in the aftermath of the conflict, multi-national relief efforts to help Kurdish refugees in Northern Iraq depended on helicopters to fly in supplies to remote mountain camps.

In the post-Cold War world, humanitarian aid and peacekeeping missions have seen ostensibly military helicopters put to extensive use. Media images of United Nations relief operations in Somalia, Haiti, Rwanda, Bosnia and elsewhere are dominated by swarms of helicopters. NATO peace enforcing missions in Bosnia have seen the Apache

attack helicopter intimidating local forces into keeping the peace.

From a communist standpoint, Soviet forces used helicopter gunships to great effect during their long conflict in Afghanistan from 1979 onwards. The simple, but rugged, Mil Mi-8 and Mi-24 assault helicopters became familiar images on snatched footage shot by western television crews cheering with the Mujahideen guerrillas at the bottom of parched Afghan valleys. In 1994 Russian helicopters were again in action against Islamic guerrillas in Chechnya. Combat helicopters from the former Soviet Union are much in demand because of their low cost and reliability. Proof of this has come very recently when, in a remarkable vote of faith in their former opponents' equipment, the Smith African-based mercenary organisation Executive Outcomes has become a regular user of Mi-8 and Mi-24s during its operations in Angola and Sierra Leone.

This book's guide aims to describe the major combat helicopters in service today, or in the final stages of development. We have classified combat helicopters as rotary-winged aircraft designed specifically for military use, or civilian machines adapted for use by military forces.

Increasingly, armed forces are making use of chartered civilian helicopters as a means to cut costs, and we have enclosed the types used by contractors in this study, particularly those chartered by the United Nations for humanitarian and peacekeeping missions.

We also take note of a number of major changes in the

helicopter industry. For example, the consolidation of helicopter manufacturers into a smaller number of larger companies is reflected in the usage of new company titles. We have, however, included details of what are termed 'heritage companies' for reference. As a rule, we have used the current name of the manufacturer, or last name manufacturer before production ceased.

The opening up of the Russian defence industry since the demise of the Soviet Union has meant that it is now possible to attribute long-established designs to their real manufacturers, rather than just link products to design bureau (known as OKB). Actual Russian helicopter and weapon designations are also used to supplement NATO reporting names.

Helicopter production continues around the world in large numbers in spite of the general down turn in global defence spending. This trend will continue as combat helicopters continue to be in the forefront of military thinking and actual operations well into the 21st Century. New technical developments such as the introduction of tilt rotors and advanced compound helicopters also offer military helicopter users significant improvements in both performance and operational capabilities.

Tim Ripley

Aerospatiale Alouette II (France)

Type: Light helicopter

Accommodation: One pilot; four passengers

Development/History

After it first flew in 1955, the Alouette II became the world's first turboshaft-powered helicopter to enter production.

Variants

SE 3130: first prototype Alouette II, powered by the 200 kW (268 shp) Turbomeca Arrius 1 turboshaft.

SE 3130B: Designation after Sud-Est merged with Gose Aviation in 1956, later renamed Sud-Aviation.

SE 3140: Alouette II development powered by a 208 kW (280 shp) Turbomeca Turmo II engine, but never produced.

SA 3180: Alouette II derivative powered by the more economical Arrius 1B with a new centrifugal clutch.

SA 3180C: Production version of SA 3180.

SA 3180L: Later: Powered by Turbomeca Arrius 1B.

Assembled in India (Cheetah) and Brazil (SA 3180Govero).

Status

French production ended 1975, Indian production continues.

Operators

Argentina, Belgium, Brazil, Bolivia, Cambodia, Chile, Congo, Ecuador, El Salvador, Dominican Republic, Germany, Guatemala, India, Lebanon, Morocco, Senegal, Spain, United States

Manufacturers

Sud-Est/Sud-Aviation/Aerospaciale (France), Hindustan Aeronautics Ltd (India), Helibras (Brazil), Saab (Sweden), Repsolde Aviacion (USA)



Aerospatiale Alouette II

(See Ripley)

Specifications (for SA 318C)

Powerplant

One Turbomeca Arrius 1B turboshaft

Power: 139 shp (200 kW) de-rated to 100 shp (75kW)

Weights

Empty: 1561 lb (708 kg)

Max takeoff: 3638 lb (1650 kg)

Payload: 1333 lb (605 kg)

Dimensions

Length: 29 ft 8 in (12.0 m)

Rotor diameter: 33 ft 6 in (10.20 m)

Height: 9 ft (2.75 m)

Performance

Max speed: 127 mph (205 km/h)

Range: 380 mi (610 km)

Armament

16 IR and 17 wire-guided anti-tank missiles, free-fight tickets, machine gun

Aerospatiale Alouette III (France)

Type: Light helicopter Accommodation: Two pilots, five passengers

Development/History

The best-selling Alouette III was one of the smaller Alouette II. The first prototype flew in 1955 and rapidly became a best-selling machine with 2,362 built and 74 countries operating the helicopter at the height of its popularity. Originally intended for service with the French armed forces in Algeria, that conflict came to an end before it was in widespread use. Portuguese, Rhodesian and South African forces used the helicopter extensively in their long conflicts with Nationalist guerrillas throughout Southern Africa. It has been used extensively in conflicts on the Indian sub-continent by Indian and Pakistani forces, performing well in the high Himalayan. Versions have been used for liaison, observation, attack, assault transport, anti-submarine warfare, anti-venture warfare, anti-aircraft, counter-attack and rescue, counter-insurgency and armed reconnaissance work.

Sud-Aviation, later Aerospatiale, were keen to use license production deals to foster business relationships in the Eastern Bloc and Third World. They were one of the first western aviation companies to offer technology transfer and work on the Alouette family helped establish the Indian, Romanian and South African helicopter industries.

Variants

SE 3160: Alouette III powered by one 649 kW (870 shp) Turbomeca Arriete 10B turboshaft, de-rated to 610 kW (816 shp)

SA 316A: Production version of SE 3160

SA 316B: Featured strengthened main and tail rotor to allow



Aerospatiale Alouette III

(Tim Ripley)

Specifications (for SA 319B)

Powerplant

One *Arriete* XV turboshaft

Power: 610 shp (649 kW) de-rated to 560 shp (747 kW)

Max T/O: 4695 lb (2100 kg)

Payload: 1650 lb (750 kg)

Performance

Max speed: 136 mph (220 km/h)

Rang: 335 mi (540 km)

Dimensions

Length: 33 ft 4 in (10.2 m)

Rotor diameter: 36 ft 1 in (11 m)

Height: 9 ft 9 in (2.9 m)

Armament

AS12 guided missiles, Mk 44 ASW torpedoes, machine guns (pod or door mounted), in-flight rocket pods.

Weights

Empty: 3436 lb (1560 kg)



Aerospatiale Alouette III of Royal Netherlands Air Force

(Tim Flerty)

for greater performance. Produced in France as SA 316B and in India as Chetak.

SA 316C: Alouette III powered version built in limited numbers.

SA 316B: Naval development of the SA 316B, powered by a more efficient and more

economical 649 kW (870 shp) Turbomeca Astazou 30V turboshaft, delayed to 447 kW(600 shp).

G-Cat: Rheinmett Air Force gunship version with ten side-mounted Browning machine guns. Gunship with single port being 20 mm

Machine gun in cabin known as E-Cat.

SA-317 Styx: Prototype Romeo gunship version, armed with anti-tank missiles, free-fall bombs and machine gun pods which did not enter production.

Atlas Aviation XM-1 Alpha, South Africa weapon system demonstrator for Russell attack helicopter.

Status

Production in France ceased in 1980 after 1455 built. Some 200 built in France until 1969. Limited production continued only in India, with 200 built to date.

Operators

Algeria, Angola, Argentina (navy), Austria, Belgium (navy), Bolivia (air), Botswana, Cambodia, Chad, Congo (Republic), Ecuador (air force), Equatorial Guinea, France (Army/Army/air force), Ghana, Greece (navy), India, Indonesia, Israel, Germany (navy), Iraq, Ireland, Jordan, Lebanon, Libya, Maldives (navy), Mexico (navy), Morocco, Mauritania, Myanmar, Namibia, Nepal, Netherlands, Norway, Pakistan (Army/Army/air force), Peru (Army/Army/air force), Portugal, Romania, Rwanda, South Africa, Tanzania, Switzerland, Tonga, USA (Mar. (USMC)), Venezuela (air force), Congo (former Zaire) and Zimbabwe.

Manufacturers

Aviation Aéronautique (France), ICA Bréguet (France), Federal Aircraft Factory (West Germany) and Hendon Aerodrome (UK/India).

Aerospatiale Super Frelon (France)

Type: Heavy lift helicopter

Accommodation: Two pilots, up to 37 passengers

Development/History

First flown in the 1962 to meet French Navy requirements for a maritime helicopter armed with anti-ship guided missiles and ASW weapons. Some remain in French service for logistic support and vertical replenishment at sea.

Variants

SA 321: Pre-production aircraft

SA 321B: French ASW version, later state to the Exocet

SA 321Bac: French navy cargo-carrying and assault transport.

SA 321BMM: Export version for Libya.

SA 321C: Civilian version.

SA 321CH: Version sold to Iraq with Eurocopter engines, Dassault DR.3000 radar and Exocet missiles.

SA 321C II: Civilian version.

SA 321C: Export transport version to Israel

SA 321C: Export transport version to South Africa.

SA 321CMM: Export transport/civilian version to Libya.

Chengdu Z-8: Chinese-built naval and combat version.

Status

Production continues in China only.

Operators

France (navy), China (navy), Iraq, and Libya.

Manufacturer

Safran Helicopters/Aerospatiale (France) and Chengdu Aircraft Industry (China).



Aerospatiale SA 321 Super Frelon

(Tim Ripley)

Specifications (for SA321G)

Powerplant

Three Turbomeca Turmo 10C turboshafts

Power: 4,710 shp (35,600 kW)

Weights

Empty: 15,130 lb (6860 kg)

Max (ZG): 29,650 lb (13,400 kg)

Payload: 11,423 lb (5000 kg)

Dimensions

Length: 63 ft 7 in (19.4 m)

Rotor diameter: 62 ft (18.9 m)

Height: 16 ft 2 in (4.9 m)

Performance

Max speed: 171 mph (275 km/h)

Range: 549 nm (1000 km)

Armament

ASW torpedoes, depth charges; machine guns

Eurocopter Gazelle (France)

Type: Light helicopter

Accommodation: One pilot, four passengers

Development/History

SAF-Aviation began work on the Gazelle in the mid-1960s as a replacement for its Alouette family. By 1967 it had been pulled into the resulting pot of the Anglo-French Helicopter Agreement, which was to set the joint development of the Gazelle, Lynx and Puma families of helicopters by Sud-Aviation (later Avionsud) and Britain's Westland. The agreement gave France the lead in export exports, and Avionsud was soon leading a major foreign sales drive. Exports and co-production deals resulted in more than 400 sales, 204 being ordered for conversion in Britain (all except 12 for the UK armed forces) while France bought just under 400. Total production was some 1,250.

A year later the Gazelle made its first flight, and won the version with the revolutionary 'Fenestron', or tail-in-the-tail, rotor system. By the mid-1980s the aircraft was in widespread use with the British and French armed forces. From 1973 the French began to field the SA 342 version, which sported an improved engine. Britain chose not to adopt the new engine for its Gazelles.

British versions saw combat in the Falklands in 1982, but it was in the 1982 Lebanon war that a Syrian version armed with HOT anti-tank missiles showed the Gazelle's true potential as an armed helicopter. French SA 316 and British anti-armoured versions were used extensively during the 1991 Gulf War in the air cavalry role on the extreme left flank of The Coalition forces. The Kuwaiti Guards fought alongside US Marine Corps forces to liberate Kuwait City. Vegetation-hunting versions have been used extensively in aerial and unarmed roles during the civil war that broke out



Westland Gazelle AH Mk 1 of the British Army Air Corps

(Tim Ripley)

Specifications (for SA 341)

Powerplant

One Turbomeca Astazou IIIA turboshaft
Rotors: 50 ft 6 in (15.3 m)

Performance

Max speed: 193 mph (310 km/h)
Range: 241 mi (390 km), 193 mi (310 km)
with max payload

Dimensions

Length: 39 ft 3 in (11.9 m)
Rotor diameter: 34 ft 5 in (10.5 m)
Height: 10 ft 2 in (3.2 m)

Armament

AS.11, AS.12, HOT, RSC and 90x140 mm
PA-3 Sagard wire-guided anti-tank missiles,
90x130 mm Shtora (SA-7) heat and thermal anti-air
missiles; Dual M60-1 20 mm rotating gun
and pod-mounted machine guns, four flight
merchants

Weights

Empty: 2,022 lb (917 kg)
Max TOW: 3,950 lb (1,800 kg)
Payload: 1,540 lb (700 kg)

In 1991, with Serb-operated Gazelles seeing action against Slovenian, Croat and Bosnian forces.

British and French Gazelles have been used in the former Yugoslavia to support United Nations and UNPROFOR peacekeeping forces since 1992. British Army Gazelles operating in Northern Ireland have been fitted with a variety of specialist observation and close circuit television systems.

Variants

SA 340: Two-seat prototype, first with conventional rotors and 4-tail, second fitted with digital main rotors and 3-tail rotor.

Attributed 8 powerplant of 262 kW (360 shp).

SA 341: Four-pax-production helicopter with enlarged cabin, semi-articulated rotors, 440 kW (590 shp) Astazou III and 2862 lb (1300 kg) maximum weight.

SA 341B: British Army Air Corps Gazelle AH 1, 212 built.

SA 340C: British Royal Navy Fleet Air Arm Gazelle HT 2, 40 built.

SA 340D: British Royal Air Force Gazelle HT 3, 29 built.

SA 340E: British Royal Air Force VIP transport Gazelle HCC 4, one built and three converted from HT 3s.

SA 340F: Civilian.

SA 340G: French Army Aviation version, 165 built.

SA 341F/Cougar: French Army Aviation MEDEVAC medevac armed version, 12 converted from original HT.

SA 341H: Initial French military export version.

SA 341H Particulier: Yvesair-built version.

SA 341M: French Army Aviation HCC armed version, 40 converted from original HT.

SA 342F: Civilian.



Eurocopter SA 342 L1 Gazelle

(Aerospaciale)



SA 342K: Up-rated military export version with 850 kW (1000 hp) Astazou XM1 Powerplant and 4100 lb (1860 kg) maximum weight.

SA 342L: Military export model with improved features. Some 170 built in Yugoslavia, including SA 342L GAMMA attack and SA 342L2 HELBA search versions armed with Soviet 82mm anti-armour and air-to-air missiles.

SA 342L1: Military export version with Astazou XM1 and 4400 lb (2000 kg) maximum weight.

SA 342M Whisler: Final production version for French Army Aviation, with Astazou XM1 and HOT missiles. More than 200 produced. Some 30 fitted with blade missile to SA 142M/Cobra standard and later SA 342M/M1 standard.

Status

No longer in production

Operators

Angola, Bosnian Serb Republic, Burundi, Cameroon, Croatia, Cyprus, Ecuador, Egypt, France (Army) Greek Republic, Iraq, Israel, Jordan, Kenya, Kuwait, Lebanon, Libya, Mexico, Qatar, Senegal, Slovenia, Syria, Tunisia, UAE (Abu Dhabi), United Kingdom (Army), Yugoslavia (Serbia/Montenegro)

Manufacturer

Sud-Aviation (Aerospatiale) Eurocopter (France), Westland Helicopters (UK), SOKO (Yugoslavia), Auto-British Helicopters Company (Egypt)

Above:
Eurocopter SA 342M Gazelle for French Army Aviation
(Tim Ryley)

Right:
Eurocopter SA 342 Gazelle fires a HOT wire guided anti-tank missile
(Aerospatiale)



Eurocopter Dauphin/Panther (France)

Type: Light helicopter

Accommodation: Two pilots, 10 troops

Development/History

Agusta began development of the Dauphin (Dauphin) as a replacement to the Alouette III in the early 1970s, with the first flight taking place in 1972. The twin-engined version first flew three years later, and it has remained in production ever since, with worldwide sales and a number of license production agreements being reached. A version with Allison engines entered service with the US Coast Guard in 1987 after a lengthy programme to integrate the US-sourced powerplant. Some have since been passed on to Israel. From 1986 onwards, military versions have been designated the Panther, with designations on the AS5 series adopted simultaneously. The Dauphin/Panther has proven to be a versatile and reliable light helicopter, which looks set to remain in production and service until well into the next century.

Variants

AS 360: Initial prototype powered by single Turbomeca Astazou 300A powerplant.

AS 360H: Initial military version powered by single Turbomeca Astazou 300HA powerplant.

AS 365C Dauphin 2: Twin-engined version powered by 405 shp (300 kw) Turbomeca Arriel turboshafts. In 1986 re-designated as AS 365C2 Dauphin 2, C1/C2, C7 versions.

AS 365H: Improved version with retractable anti-birdstrike.

AS 365H1: Further improvement with 31-knuckled transmission and up-rated Arriel 101 powerplant.

AS 365H2: Civil version with Turbomeca 1C9 powerplants.

AS 365H3: First military version of twin-engined



Eurocopter AS 365H3 Dauphin

NDP Spokesman

Specifications (for AS 565 Panther)

Powerplant

Two Turbomeca Arriel 101 turboshafts

Power: 1500 shp (1120 kw)

Payload: 2527 lb (1150 kg)

Dimensions

Length: 38 ft 1 in (11.6 m)

Rotor diameter: 39 ft 8 in (12.1 m)

Height: 13 ft 1 in (3.99 m)

Performance

Max speed: 104 mph (166 kmh)

Range: 427 nm (775 km)

Weights

Empty: 4825 lb (2190 kg)

Max T/O: 9300 lb (4220 kg)

Armament

Gat M621 20 mm canon pods; Matra air-to-air missiles; HOT wire-guided anti-tank missiles; Free-fight rockets; AGM-114 and Exocet anti-ship missiles; Murray torpedoes.

AS365D2, for attack and troop transport. This was renamed the Panther, powered by Turbomeca 101T and materialized under the following versions: AS 365M1 fire-flight rocket and gun armed; AS 365M2 utility, AS 365N1 anti-tank; AS 365F navalized version with retractable undercarriage; AS 365F1 navalized version; AS 365SA anti-shipping; AS 365MA unarmed version; AS 365N2 anti-Amba rescue.

AS 365N3: Upgraded version with two Turbomeca Arriel 2C turboshafts. Panther versions were designated SA 365N3 utility; SA 365N3M1 countermeasures armed; SA 365N3M2反潜 utility; SA 365N3M3夜战夜航 armed.

AS 365N4: Owl side look version, seating 14 and powered by Arriel 2C.

AS 365 Panther: Brazilian version of 8 model designated HM-1 by Brazilian army.

Panther 600: Proposed US Army version. Did not enter production.

AS 365D3: Version produced for US Coast Guard under designation HH-65A, with Britain Lynxwing UH-1-700A-1 engines, specialist night vision and rescue equipment. Also used by Israel.

Harbin Z-9 Helian: Chinese version assembled from French kits.

Harbin Z-9A-100: Chinese-made version with WZ-10A powerplant, rated to 540 kW (735 shp), which can be armed.

Status

In production in France, Brazil and China.



Eurocopter AS 365 Panther

(Aeroportal)



Operators

Angola, Bangladesh, Brazil (Army), Burkina Faso, Cambodia, China, Costa Rica, Dominican Republic, Egypt, France (Army, Air Force), India (Air Force), Ireland, Israel, Romania, Saudi Arabia (Army), Sri Lanka, Taiwan, Thailand (Army), UAE (Air Force), USA (Coast Guard)

Manufacturers

AleniaAeritalia (Italy), Helicor (Brazil) and Helicor Aircraft Manufacturing (Korea)

**Eurocopter AS 565F
Panther**
(Tim Ryley)

Eurocopter Ecureuil/Fennec (France)

Type: Light helicopter

Accommodation: Two pilots, two/three passengers

Development/History

The three-seat testbed Ecureuil first flew in 1974 and has remained in production ever since, attracting several thousand military and civilian customers around the globe. The single-engined 260 series version has been supplemented by the twin-engined 350 series aircraft, which provided greater performance. Since 1996 specialist military versions of the Ecureuil have been marketed under the former Dassault name, using the newer 355 series designation.

Versions

AS 350 Ecureuil: First prototype with single engine

Lycoming T53-101 turboshaft

AS 350BA/B2/B3: Civilian/military version with single

Lycoming Aviadv. 1B, B2 with Aviadv. 1D1; B3 with Aviadv. 2.

AS 350B: Civilian version with single engine Lycoming T53

HT turboshaft. Known as AS350 in USA

AS 350 Firefighter: Specialised version

AS 350B2: First armed version, powered by 600 kW (770

shp) Turbomeca Arriel 101, known as Fennec. AS 350B2/C3

anti-tank version. Other Fennec versions include: AS

350B2/C3 unarmed utility; AS 350B2/C3 armed, canister or

rockets; AS 350B2/C3 unarmed naval; AS 350B2 armed naval

anti-submarine; AS 350B2/C3 ar Aviadv. 2B powered.

BR 350B1/11 Esquilo: Unarmed Brazilian version, designated

CH-90 and TH-90 by Brazilian Air Force, CH-12 by

Brazilian Navy

BR 350A1 Esquilo: Armed Brazilian version, designated

EA-1 by Brazilian army.

Squadron HT 1/HT 2: UK training version of AS 350B.



Eurocopter AS 355 Ecureuil

(Tim Ripley)

Specifications (for AS 350B)

Powerplant

One Turbomeca Aviadv. 1B turboshaft

Power: 601 shp (450 kW)

Dimensions

Length: 36 ft 10 in (11.1 m)

Rotor diameter: 35 ft (10.7 m)

Height: 10 ft 11 in (3.3 m)

Weights

Empty: 2325 lb (1050 kg)

Max UD: 4620 lb (2100 kg)

Performance

Max speed: 178 mph (287 kmh)

Range: 334 nm (300 km)

Armament

One M621 20 mm-cannon pod; 762 mm

machine gun pod; four-light mortars; 1000

wire-guided anti-tank missiles; Matra air-to-

air missiles; anti-submarine torpedoes



AS 350BA in service with the Australian Army
IAFW

AS 350 Ecureuil: first production version with two 313 kW/420 shp Alvis 250-C20F turboshafts.
AS 350M Improved: improved version with two 340 kW/455 shp Turbomeca Arrius 1A. Civil version known as AS 350F. Also Star in USA.
AS 350F: improved rotor blade version.

AS 350F1: French training version. F2 has upgraded transducers.

AS 350M2: French armed version.

AS 355 Fennec: twin-engined version. AS 355M armed version with 30 mm canon; AS 355M1 training and utility version; AS 355M2 armed naval version; AS 355M3 command/transport version; AS 355M4 utility version; AS 355M5 naval utility version; AS 355M6 unarmed naval version; AS 355M7 armed naval version.

Z-11: Chinese produced copy with YUZ-8D Powerplant, rated to 310 kW/414 shp.

AS/HB 355E2: Bearcat version, designated CH-35 and VH-35, of Lynx Mk. 8, by Australian Air Force, VH-128 by Bearcat navy.

Twin Squirrel: UK MP transport version of AS 350M1.

Status

In production at France, China and Brazil.

Operators

Argentina (coast guard), Australia (army, navy, air force), Brazil, Botswana, Brazil (army, navy air force), Central African Republic, Denmark (army), Ethiopia, Ecuador (army), Eq. France (army, navy, air force), Ireland, Ireland, Malawi, Paraguay, Peru (air force), Sierra Leone, Singapore, Slovenia, USA (Air Force, US Air Force, army).

Manufacturers

Amcorp (China), Agusta (Italy), Chang (China) and Helibras (Brazil).



Eurocopter AS 350 CS Fennec

(Eurocopter)

Eurocopter Puma (France)

Type: Medium lift helicopter

Accommodation: Two pilots, loadmaster, 20 troops

Development/History

Work on the SA 330 began in 1963 but the programme became multi-national as a result of the 1967 Anglo-French helicopter agreement. This resulted in Westland building all for the British Royal Air Force. Under this arrangement future development and export work on the design was the responsibility of Aérospatiale, later Eurocopter, who began a vigorous sales drive in the 1970s. British and French Pumas have seen service in the 1991 Gulf War and supporting peacekeeping missions in the former Yugoslavia. South African forces used the Puma extensively in their bush wars in Angola and South West Africa. French production ceased in 1987 after 287 built. The design was superseded by Super Puma (Ovogat version) from the late 1980s. The main centres of Puma development are now in South Africa (see Ovogat entry) and Romania, where extensively upgraded versions are produced.

Variants

SA 330B: First eight French prototypes.

SA 330B2: French Army Aviation version.

SA 330C: Military transport version.

SA 330E: Royal Air Force version, designated Puma HC.1.

SA 330E BG: Croatian version with 1024 kW (1375 shp)透博梅卡MT powerplant.

SA 330E: Military version with 1174 kW (1575 shp)透博梅卡 powerplant. Designated SA 330E by French air force, even though different from the French army's SA 330B. SA 330U H L: Upgraded C H H version with glass-fibre rotor blades.



Westland Puma HC.Mk 1

(Tom Ryley)

Specifications (for SA 330)

Powerplant

Two Turbomeca Turmo HC turboshafts

Power: 2150 shp (2550 kW)

Dimensions

Length: 46 ft 1 in (14.1 m)

Rotor diameter: 49 ft 2 in (15 m)

Height: 16 ft 10 in (5.1 m)

Weights

Empty: 8300 lb (3760 kg)

Max T.O. 16,315 lb (7400 kg)

Payload: 10,550 lb (4790 kg)

Performance

Max speed: 161 mph (261 kmh)

Range: 309 nm (562 km)

Armament

Machine guns; Romanian version equipped for flight rocket pods; MILAV Malyutka (AT-3 "Kappa") wire-guided anti-tank missiles. Hull-mounted 20 mm cannon, 220-lb (100 kg) free-fall bombs, A-60 air-to-air missiles.

SA 330B: Post-qual versions with OHB 31 radar for maritime surveillance and Merlin powerplant.
SA 330B: Romanian-built version. Systems upgrade underway including installation of SOCOM (Opto-ic Search and Combat Anti-Tank) weapon package.
Puma 2000: Proposed Romanian version with glass cockpit.
HS 330: Indian surveillance version.
AS 330B (Orchestr): Experimental French test bed for Orchestr ground surveillance radar.

Status

Production continues only in Romania.

Operators

Argentina (coast guard/camp), Algeria, Cameroon, Chile (Army), Congo (DRC), Côte d'Ivoire, Ecuador, Ethiopia, France (Army/Army Israel), Gabon, Guinea Republic, Indonesia (air force), Iraq, Kenya, Kuwait, Lebanon, Libya, Morocco, Nepal, Nigeria, Pakistan (army/air force), Philippines, Portugal, Romania, Senegal, South Africa, Spain, Sudan, Togo, UAE (Abu Dhabi), United Kingdom (air force).

Manufacturer

Société Anonyme des Établissements (Sud-Aviation)/Aérospatiale/Eurocopter (France), Westland Helicopters (UK), HHI (Indonesia), MSA (Russia) (Russia)



Eurocopter SA 330B Puma

(Tim Ripley)

Eurocopter Super Puma/Cougar (France)

Type: Medium lift helicopter

Accommodation: Two pilots, loadmaster, 25 passengers

Development/History

A 'growth' development of the basic Puma, the Super Puma first flew in 1979 featuring more powerful M635 powerplants. Although aimed mainly at the civilian market, Avionspatiale (now Eurocopter) have marketed specific military versions under the brand name Cougar, using the earlier AS332 designations. Stretched versions with greater seating capacity have been fielded, and a wide range of armament options are available. Recent developments have included a number of night vision options and in-flight refuelling for combat-search and rescue. The French Army are still planning to use the Cougar as the platform for their RORO/OPF ground surveillance radar system.

Variants

AS 332M1: First military version with M635 powerplants.

AS 332C: First civil version.

AS 332F1: Naval version.

AS 332L1: 'Stretched' civilian version.

AS 332L2 Super Puma Mk 2: Civil transport.

AS 332L2 Super Puma Mk 2 VIP: Civil VIP transport.

AS 332M: 'Stretched' military version, production ceased.

AS 332M1: 'Stretched' military version.

AS 332 Cougar Mk 1: In 1980 B, I and II versions re-designated and the name Cougar adopted for military sales.

AS 332MC, 1B and 1C for short-haul and military transport/assault; AS 332M, and UL for long-haul; military armed/assault; AS 332SC naval, armed anti-submarine/anti-ship.



Eurocopter AS 332 M1 Super Puma

(Eurocopter)

Specifications (for AS 332UL Cougar Mk 1)

Powerplant

Two Turbomeca Makila 1A1 free turbines

Power: 1,744 shp (1300 kW)

Max 1/h: 15.641 lb (7000 kg)

Payload: 5020 lb (4500 kg)

Dimensions

Length: 50 ft 11 in (15.5 m)

Rotor diameter: 51 ft 2 in (15.6 m)

Height: 15 ft 9 in (4.8 m)

Max speed: 112 mph (180 km/h)

Range: 334 nm (615 km)

Weights

Empty: 13,460 lb (6110 kg)

Performance

Max speed: 112 mph (180 km/h)

Range: 334 nm (615 km)

Armament

20 mm or AG32 naval gun, free-fight rocket, naval versions can carry the AS.30 laser-activated missile or homing torpedo.



Eurocopter AS 332 UL Cougar Mk. 1 with Herion battlefield surveillance system

(Eurocopter)

AS 332 Cougar Mk. 2: stretched version with 1600 kW (2100 shp) Makila 1A2 powerplant. Civilian counterpart designated Super Puma II. In-flight refueling option. AS 332M2 armed combat search version. AS 332U2 unarmed utility with stretched fuselage. AS 332M naval, armed anti-submarine.

Cougar 100: Reduced capability export version.

AS 332UL HOMELAND: Broad surveillance version developed from the halic system.

MAS 332B: Indonesian utility designation.

MAS 332C: Indonesian naval designation.

CH-34: Brazilian designation for 332M.

HL.121: Spanish Army designation for 332B.

HO.21: Spanish Air Force search and rescue designation.

HO.21/H: Spanish VIP designation.

Hkp.10: Swedish search and rescue designation.

Status

In production in France and Indonesia.

Operators

Australia (land guard, 2002). Brazil (air/sea force), Canada, Chile (air/sea/guard force), China, Congo (air), Costa Rica, Ecuador, France (air force/army), Gabon, Indonesia (air/sea force), Iraq, Japan, Jordan, Kuwait, Malaysia, Mexico (air force), Nepal, Netherlands, Nigeria, Panama, Peru (air/sea), Qatar, Saudi Arabia (air/sea/guard force), Singapore, South Korea (air force), Spain (army), Sweden (air force), Switzerland, Thailand (air force/sea force), Turkey (army), UAE (Abu Dhabi), Venezuela.



Manufacturer

Aerogust/Atalanta/Eurocopter (France), IPTN (Indonesia) and Singapore aircraft were assembled in country from kits. TH (Turkey) has signed a deal for co-production.

**Eurocopter AS 332C
Cougar**
(Eurocopter)



Eurocopter AS 332A2 Cougar rescue version with in-flight refuelling probe

(Tim Ripley)

Eurocopter BO 105 (Germany)

Type: Light helicopter

Accommodations: Two pilots, three passengers

Development/History

The German light helicopter made its first flight in 1962, and by the mid-1970s was in widespread service with the German Army - some 96 light observation and 200 BO105 attack/armed anti-tank versions were eventually delivered. Delays in the French-German light programme meant it will have to soldier on in these roles until well into the next decade. It has been widely exported to civilian and military customers around the world.

Variants

BO 105C: Initial version.

BO 105CB: Basic light observation/utility version.

BO 105CBS: Stretched utility version, with capacity for five passengers.

BO 105CBS: Stretched utility version, with capacity for six passengers. Renamed H1 by Swedish Army.

BO 105G2: Canadian produced version with up-rated Allison 250-C20C powerplant.

BO 105G4 (WBM): German road version.

BO 105GVM-1: Basic German anti-armour version fitted with six HOT missile tubes.

BO 105GVM-1A1: Improved German anti-armour version with new rotors.

BO 105GVM-1 Phase 2: Proposed German night attack version.

BO 105GVM: Proposed German export version with four Stinger air-to-air missiles.

BO 105Gophic: Irish aircraft with mast-mounted night

BO 105GVM/MA.1G: Spanish anti-armour version.



Eurocopter BO 105 GVM

Eurocopter

Specifications (for BO 105C)

Powerplant

Two Allison 250-C20B turboshafts

Power: 640 shp (626 kW)

Max T/O: 5,511 lb (2,500 kg)

Payload: n/a

Performance

Max speed: 140 mph (220 kmh)

Rang: 550 nm (1000 km)

Dimensions

Length: 20 ft 11 in (6.3 m)

Rotor diameter: 33 ft 9 in (10.3 m)

Height: 9 ft 11 in (3 m)

Weights

Empty: 2,830 lb (1,300 kg)

Armament

HOT and RKT wire-guided anti-tank missiles,
20 mm Rheinmetall cannon

BO-105CBM/HG/A-3: Spanish armed coast version with 20 mm cannons.

BO-105CBM/HG/B-3: Spanish observation version.

MBB BO-105: West German-built version.

HBO-105S: Swedish Indonesian version.

BF-105CB5-SM455: Search and rescue/maritime version with searchlight cellar.

BO-105 LS A-3: Powered by two Allison 250-C 28C engines. Super litter, optimised for under-slung loads.

EC-super line: High performance version of CBS for civil market.

Status

In production.

Operators

Albania, Brazil, Chile (air force), Chile, Colombia (Army), Germany (Army), Indonesia (Army/Naval Air Force), Iraq, Jordan, Kenya, Lesotho, Mexico (Army), Netherlands (Army), Nigeria, Peru (Army), Philippines (Army), Spain (Army), Sweden (Army), Thailand, USA (DOD).

Manufacturer

Messerschmitt-Bölkow-Blohm/Eurocopter (Germany), IHP (Indonesia), OMA (Brazil), Eurocopter Canada (Canada).



Eurocopter Tiger (International)

Type: Attack helicopter

Accommodation: Pilot (front), weapons operator (rear) in tandem

Development/History

Intended to replace the Gazelle in French service and the BO 105 in German service, the Tiger has its origins in a memorandum of understanding signed by the two countries in 1984. After a protracted process, a development contract was signed in November 1993 and work began in earnest to produce five prototypes.

In the early years of the programme both France and Germany were keen supporters of the Tiger, but delivery setbacks in the 1990s have forced the delivery programme to be stretched out, with the first batch of 80 aircraft for each country not entering service until the next century (Germany in 2001 and France in 2003). Initially Germany will receive only AHU close support versions, while the French are to receive 20 anti-tank support and 18 anti-tank models. Production of the remaining aircraft will then last until 2025, with a total of 216 being built for France and 212 for Germany.

Anti-tank versions are armed with HOT or TOW anti-tank missiles, a mast-mounted forward looking infra-red sight and air-to-air missiles are also optional. The anti-air support versions are armed with a barrel-mounted 30 mm GAU cannon under the nose, air-to-air missiles and machine gun pods.

Variants

HAP (Hawk): Initial German escort version.

HAP: French escort version.

HAC-1 (Acer): French anti-tank version.

PAN-2 (Tiger): Initial German anti-tank version.



(Eurocopter)

Eurocopter Tiger

Specifications

Powerplant

Two M100/Haefl-Royce/Turbomeca M112 250 turboshafts

Power: 2370 shp (1766 kW)

Dimensions

Length: 46 ft 11 in (14.3 m)

Rotor diameter: 42 ft 2 in (12.8 m)

Height: 14 ft 2 in (4.3 m)

Weights

Empty: 7275 lb (3299 kg)

Max (OF): 12,287 lb (5580 kg)

Performance

Max speed: 174 mph (280 km/h)

Endurance: 2 hours 50 min

Armament

(HAP) GAU-30/31 30 mm cannon, M112 air-to-air missiles, 68 mm rockets; (PAN-2) HOT 2D wire-guided anti-tank missiles; long-range laser-guided anti-tank missiles, AGM-114 laser-guided anti-tank missiles, Stinger or M113 air-to-air missiles, machine gun pods

Eurocopter Tiger (International)

(UH): General multi-role close support version, originally designated UH-1.
HCP: Export multi-role version, without roof-mounted sight.
U-Tiger: Export anti-tank version.

Status

In pre-production.

Operators

84

Manufacturer

Eurocopter (France/Germany)



*Eurocopter Tiger
(Eurocopter)*

EH Industries EH.101 Merlin (International)

Type: Shipborne ASW helicopter/utility helicopter

Accommodation: Two pilots, observer, sonar operator

Development/History

The joint British-Italian collaborative programme began in 1979 to develop a Sea King replacement for both countries' navies. Funding was agreed in 1984 to proceed with building nine prototypes and subsequent development. The first prototype flew in the UK in 1987, and since then, the programme has led to the development of dedicated maritime, utility, airborne early warning and civil passenger versions. Current order books stand at 44 maritime versions for the British Royal Navy and 32 utility for the Royal Air Force in Wales and Puma replacement. Italy's Navy has ordered eight maritime, four airborne early warning and four utility versions.

Major orders were expected from Canada but the programme was cancelled in 1993 after a change of government. Export orders now being keenly sought from Canada (Argus), Portugal, Japan and the Middle East. The Merlin programme for the Royal Navy is unique because Westland - the airframe manufacturer - is not the prime contractor. Lockheed Martin is prime contractor, being responsible for integrating the complex anti-submarine sensor and weapon systems with the airframe.

Variants

Merlin Mk. 1: Royal Navy maritime helicopter.

EH.101 ASW/ASW/AM: Italian maritime helicopter.

EH.101 AW: Italian airborne early warning version.

EH.101 Utility: Italian naval transport version.

Merlin HC.3: RAF support helicopter.

Merlin: Civilian version.



EH Industries EH.101 Merlin

AGM Westland

Specifications (Basic Naval version)

Powerplant

Two Rolls-Royce Turbomeca RTM 322

turboshafts (UK); General Electric T700-GE-10A (Italy)

Power: 6300 shp (3172 kW) - 5942 shp (3634 kW)

Weights

Empty: 15 700 lb (7129 kg)

Max T/O: 38 640 lb (17 500 kg)

Payload: 8 000 lb (3630 kg)

Performance

Max speed: 150 mph (245 km/h)

Range: 625 nm (1118 km)

Armament

Mk 46, Sting Ray torpedoes; Sea Skua anti-guided anti-aircraft missiles, depth charges.



CH-148 Fleet: Proposed Canadian maritime version.

CH-148 Chima: Proposed Canadian rescue version.

Combi: Proposed Canadian review version.

Status
In production.

Operators
Italy (navy), UK (navy/air force).

Manufacturer
Agusta (Italy) and Westland
Helicopters/GKN Westland (UK).

Left
EH Industries EH.101 Merlin
(GKN Westland)

Right
EH Industries EH.101 Merlin
(GKN Westland)



NATO Helicopter Industries NH 90 (International)

Type: Multi-role medium-lift/multimission helicopter

Development/History

The multi-national project began in 1989 and originally involved five nations. However, Britain pulled out in 1991, leaving France, Germany, Italy and the Netherlands to continue building the MATO Frigate Helicopters (NH90) and Tactical Transport Helicopter (TTH). Full scale development began in 1992, and the first prototype flew in 1995. The second prototype equipped with fly-by-wire flight control systems flew in 1997.

Delays resulting in Western Europe have led to the programme being scaled down and delivery dates slipped. In 1996-1997 the funding for the production delivery schedules was agreed. The Netherlands is taking 20 NH90 versions from 2001, Germany wants 205 tactical transports from 2001 and 28 TTHs from 2007, France has ordered 27 NH90s from 2005 and 100 TTHs from 2011, and Italy requires 80 NH90s and 14 TTHs from 2004. In total, 647 helicopters are on order, but few commentators expect the programme to survive future European defence budget cuts.

Variants

NH90: MATO Frigate: 16 versions for反-submarine and utility tasks.

TTH: Tactical Transport Helicopter

Status

In production.

Operators

None

Accommodation: Two pilots, (NH9) three systems operators, (TTH) 20 troops



NH Industries NH 90

Specifications (For NH90)

Powerplant

Two Rolls-Royce Turbomeca/Turgo RTM 322-01/0 or General Electric/Alfa Romeo M252-401X turboshafts.

Power: Respectively 4200 shp (3100 kW) or 4600 shp (3500 kW)

Weights

Empty: 14,741 lb (6420 kg)

Max T.O. 20,047 lb (9000 kg)

Payload: 6409 lb (2900 kg)

Performance

Max speed: 164 mph (260 km/h)

Ferry range: 650 nm (1200 km)

Armament

Anti-submarine: Ingolfsen, anti-ship missiles, depth charges, 702 mm or 127 mm torp. guns

Dimensions

Length: 63 ft 10 in (19.3 m)

Rotor diameter: 53 ft 5 in (16.2 m)

Height: 13 ft 10 in (4.22 m)

Manufacturer

NH, with Eurocopter
(Germany), Agusta (Italy)
and Fokker (Netherlands)



Right:

*NH Industries NH90
(Courtesy Fokker/AFS)*

Agusta A 109 (Italy)

Type: light helicopter (for A 109CM)

Accommodation: Two pilots, six passengers

Development/History

Agusta's stylin' light helicopter first flew in 1980 and has sold well around the world since 1981. Armed military versions first entered service with the Italian Army in 1988, although Belgium is the only export customer for this model to date; more than 600 have been produced in all military and civil versions.

Variants

A 109: Initial production version

A 109A Mk II: Civil version

A 109B: "Wise-Body" version with improved transmission

A 109B04: Basic Italian army version

A 109HM: Current production military version with sensor weapon improvements

A 109HAW: Italian Army version with Helibrain wire-guided anti-tank missiles

A 109H2: Improved transmission and longer nose for more avionics

A 109H2S: Swiss export version

A 109HMM: Fixed undercarriage, with 550 kW (735 shp)

Anti-ship: Agusta A 109P: Powerplant for "Hot and High" operations

A 109HMS: Naval version

A 109MMX: Extended mission version

A 109Gulf: Coast guard version

A 109 Power: Two Pratt & Whitney 2000 powerplants, each rated to 732 shp (546 kW)

Status

In production



Italian army Agusta A 109

Library Photo/AFPS

Specifications

Powerplant

Two Alvis (Mitsubishi) turboshafts
Power: 900 shp (670 kW)

Dimensions

Length: 25 ft 8 in (7.8 m)
Rotor diameter: 36 ft 1 in (10.9 m)
Height: 11 ft 5 in (3.5 m)

Weights

Empty: 3,933 lb (1,780 kg)

Max T/O: 5,997 lb (2,720 kg)

Payload: Underwing 2,000 lb (907 kg)

Performance

Max speed: 150 mph (241 km/h)
Range: 420 nm (77 km)

Armament

TOW-2A wire-guided anti-tank missiles;
machine gun pods, free-fall rocket pods;
Stinger air-to-air missiles.

Operators

Argentina (Argentina), Belgium,
Italy (Italy), Malaysia, Peru
(Peru), Slovenia, US (Army),
Venezuela (Venezuela)

Manufacturer

Agusta (Italy).



Right

Agusta A 109 Mangusta
(Tim Ripley)

Agusta A 129 Mangusta (Italy)

Type: Light attack helicopter

Accommodation: Two pilots in tandem

Development/History

Italy's distinctive Mangusta (Mongoose) is the first custom-designed Western European attack helicopter to enter front-line service with a NATO country. With a track record in helicopter combat since dating back to 1952, Agusta began working on the Mangusta in the mid-1980s in response to an Italian Army requirement for a specialist anti-tank helicopter.

US experiments with the Cobra and early versions of the Apache obviously influenced the design of the Mangusta, which made its first flight in 1983. Five prototypes were flying by 1986, with a delivery date scheduled for the end of 1987. However, the first production aircraft were not delivered until 1990, with 15 being subsequently produced per month. The delay in delivery was due to funding problems with the Hughes/Emerico/Saab Hel-TOW rear-mounted anti-tank missile sight system.

The initial Italian Army order for 60 aircraft has since been followed by plans to develop a multi-role scout/gunship version. This variant boasts a chin-mounted twin 30-mm (1.18-in) or 12.7-mm (0.5-in) machine gun. If a new build version is not ordered, then 20 of the original airframes may be converted. Despite the A129 seeing combat service with the Italian United Nations contingent in Somalia during 1993, export orders have not been forthcoming – it has sold just in British, Dutch, Malaysian and several Middle Eastern attack helicopter computations.



Agusta A 129 Mangusta

(Ferrari Display)

Specifications

Powerplant

Two Rolls-Royce 1004D turboshafts.
Power: 1850 shp (1389 kW)

Dimensions

Length: 40 ft 2 in (12.3 m)
Rotor diameter: 39 ft (11.9 m)
Height: 11 ft (3.3 m)

Weights

Empty: 5575 lb (2529 kg)
Max T/O: 10,025 lb (4548 kg)
External loadout: 2045 lb (1200 kg)

Performance

Max speed: 160 mph (256 km/h)
Endurance: 3 hours 5 minutes

Armament

Four hard points, BGM, BOW II or 3A wire-guided anti-tank missiles; Hellfire laser-guided anti-tank missile; AGM-114 Hellfire, Stinger, Law-2, Maverick air-to-air missiles; machine gun pods; free-fight rocket pods; 20-mm (0.79-in) chin gun (can fire 1000 rounds); or 12.7-mm (0.5-in) chin gun (can be fired but not in service).

Variants

- A 129: Basic Italian Army anti-tank version.
- A 129 Sowet: Proposed reconnaissance version with mast-mounted sight and chin gun fitted.
- A 129 International: Export version with two LHTEC T800 engines, five main rotor blades and improved avionics system.
- A 129 Shipkuster: Proposed navalized version.
- A 129 Multi-Role: Proposed follow-on to current in-service version, similar in capability to International version, and armed with barrel-mounted 20 mm Giatling gun.

Status

In production.

Operators

Italy (Army)

Manufacturer

Agusta (Italy)

Right

Agusta A 129 Mangusta
(Tim Blythe)



Agusta-Bell AB 212 (Italy)

Type: shipborne anti-submarine helicopter

Development/History

This specialized anti-submarine version of the popular AB 412 helicopter has become the standard shipborne helicopter for many NATO navies. They are easily identified by the large radar housing above the cockpit and under the forward hull. A variety of surface surveillance radars have been installed, including MEL, AMI, SRS-1000, MWPS-2000 or Ferranti Scanmap. AMI/AGS-1000 dipping sonars have been fitted for anti-submarine work. All weapon carriage is external, with either a mix of anti-submarine torpedoes or anti-ship missiles. Iraqi and Iranian versions saw action during the 1980-88 Gulf War, while Italian, Greek, Spanish and Turkish versions were used to enforce UN sanctions against the former Yugoslavia.

Variants

AB 212 ASW: Basic version.

AB 212 ASW: Israeli electronic warfare version.

HA.10: Spanish designation.

Status

In production

Operators

Denmark (navy), Israel (navy), Italy (navy), Peru (navy), Spain (navy), Turkey (navy), Venezuela (navy)

Manufacturer

Agusta (Italy).

Accommodation: two pilots, sonar operator/radar operator, or seven passengers



Agusta-Bell AB 212 ASW

(Tom Ripley)

Specifications (for AB 212 ASW)

Powerplant

one Pratt & Whitney PT6A-6 Turbo Twin Pac
Power: 183% shp (1798 kW)

Max t/t: 11.176 lb (5070 kg)

Payload: 5000 lb (2270 kg)

Dimensions

Length: 42 ft 4 in (12.9 m)
Rotor diameter: 48 ft 7 in (14.7 m)
Height: 14 ft 10 in (4.5 m)

Performance

Max speed: 172 mph (156 km/h)
Range: 360 nm (667 km)

Weights

Empty: 7450 lb (3380 kg)

Armament

AS.12, Sea Killer 2, Sea Skua radar-guided anti-ship missiles, Mk 46, 66 or 100-44 torpedoes; depth charges; machine guns.

Kawasaki OH-1 (Japan)

Type: Light attack and observation helicopter

Accommodation: Pilot, gunner/observer

Development/History

The first military helicopter designed entirely in Japan is intended to replace the OH-6 in Japanese Ground Self-Defense Force service in the early part of the next century. A mock-up was revealed in 1994, and the first prototype flew two years later. Similar in appearance to the Agusta A 129, but the OH-1 features a tandem tail rotor and 1990s generation materials, avionics and weapon systems. The 1997 defense budget included funding for the first three production aircraft.

Variants

OH-1

Status

In pre-production.

Operators

NA

Manufacturer

Kawasaki and Fuji Heavy Industries (Japan)



Kawasaki OH-1

Specifications

Powerplant

Two MHI SH-10 turboshafts

Power: 1340 shp (1000 kW)

Max. takeoff weight: 7710 lb (3500 kg)

Payload: n/a

Dimensions

Length: 29 ft 4 in (8.9 m)

Rotor diameter: 37 ft 9 in (11.5 m)

Height: 12 ft 5 in (3.8 m)

Performance

Cruising speed: 161 mph (260 km/h)

Range: 124 mi (200 km)

Weights

Empty: n/a

Armament

Includes Tiger 91 air-to-air missiles, anti-tank guided missiles, free-fall bombs; laser- and push-mounted cameras/guns

PZL Swidnik W-3 Sokol (Poland)

Type: Medium-lift multi-purpose helicopter

Accommodation: Two pilots, 12 passengers

Development/History

PZL Swidnik began to work on upgrading the old W-2 design during the 1970s, and the result of that work, the W-3, began test flying in 1979. Production began in 1985, and it has since entered service with the Polish armed forces.

Development to field-armed versions is underway, with the help of South Africa and Israel, to improve the export potential of the helicopter by giving customers western and eastern weapons options.

Variants

W-3 Sokol: Standard civil and military version.

W-3M Tuszka: Streched version with up-rated engine to T450-11 (1000 shp) and capacity for 14 passengers.

W-3M4 Arkanos: Polish Navy search and rescue version.

W-3M Salamandra: Gunship version.

W-3M-1 Aligator: Proposed anti-submarine version.

W-3M: Low cost armed version for Poland.

W-3MV: Proposed naval strike version.

W-3Mz: Improved avionics version for western markets.

W-3Mz: An W-3A with flotation bags.

W-3Mz Husar: Armed version upgraded with assistance from South Africa's Denel using the HawkEye weapons system.

W-3 Salam: VIP transport.

W-3 EW: Proposed electronic warfare version.

W-3 M5/WM: Proposed gunship version with tandem cockpit.

SPH-3: Proposed up-engined with Pratt & Whitney PW206B turboshafts.

Specifications (for Sokol)

Powerplant

Two WSK-PZL Rzeczen PZL-10W turboshafts.

Power: 1000 shp (714 kW)

Dimensions

Length: 46 ft 7 in (14.2 m)

Rotor diameter: 51 ft 11 in (15.7 m)

Height: 13 ft 6 in (4.12 m)

Weights

Empty: 7275 lb (3299 kg)

Max T/O: 14 110 lb (6400 kg)

Payload: 4620 lb (2100 kg)

Performance

Max speed: 156 mph (255 km/h)

Range: 661 nm (761 km)

Armament

W-3M has 23 mm (2xh-23) cannon pods, 20 mm automatic minigun, T-13 60mm and 120mm laser-guided missiles, M1134 Shrike (PA-1) Small radius- and laser beam-guided anti-tank missiles, WMDM Strela (SA-7 Grail) anti-air missiles, free-fall rockets, mine dispensers.

Status

In production

PZL Swidnik W-3 Sokol
(Tom Ryszk)

Operators

Czech Republic, Poland (Armaghawie Rosz), Nigeria.

Manufacturer

PZL Swidnik (Poland).



Kamov Ka-25 (Russia) NATO reporting name 'Hormone'

Type: Shipborne anti-submarine helicopter

Development/History

Some 460 Ka-25s were built for service aboard Soviet Navy ships from 1966. It has now been withdrawn from Russian Navy service, but a few are operational elsewhere.

Variants

Ka-25PL: Radar version.

Ka-25M: Proposed land-based attack helicopter.

Ka-25B 'Hormone-A': Original ASW version with search radar, MAD sensor, dipping sonar and torpedo launcher.

Ka-25B 'Hormone-B': Specialized version to provide target acquisition and course guidance for submarine- and ship-launched cruise missiles. Partially retractable undercarriage.

Ka-25M, N, R: Missile tracking version.

Ka-25PS 'Hormone-C': Specialized search and rescue version, without anti-submarine warfare equipment.

Ka-25LSh: Mine warfare version.

Ka-25K: Prototype flying crane.

Status

No longer in production.

Operators

India (Army), Russia (Army), Syria (Army), Vietnam, Yugoslavia (Army)

Manufacturer

Kamov Aviacon (Russia) (Joint Russian and Ultra Unit (Russia) in Turkey) (Russia) design

Accommodation: Two pilots, [optional] 12 passengers



Ka-25B 'Hormone-A' on the Minsk

James

Specifications (for Ka-25Bsh)

Powerplant

Two M62 GTD-3F turboshafts

Power: 1276 shp (924 kW)

Weights

Empty: 10,505 lb (4765 kg)

Max TOW: 15,473 lb (7020 kg)

Dimensions

Length: 32 ft (9.7 m)

Rotor diameter: 51 ft 7 in (15.7 m)

Height: 12 ft 7 in (3.8 m)

Performance

Max speed: 110 mph (180 km/h)

Range: 251 nm (460 km) with external tanks

Armament

Anti-submarine: torpedoes, depth charges

Kamov Ka-27/28/32 (Russia) NATO reporting name 'Helix'

Type: Shipborne anti-submarine helicopter

Development/History

The Ka-27 series has a larger fuselage than the Ka-25. The first prototype flew in 1974 and it entered front-line service with the Soviet Navy in the early 1980s. Its robust design and rugged construction have proven popular with crews.

Variants

Ka-27P: 'Helix-A': Basic version for Soviet Navy, also known as Ka-25BPA.

Ka-27PS 'Helix-B': Naval search and rescue version.

Ka-27PV: Armed version of PS.

Ka-28 'Helix-C': Export version of PS.

Ka-32 'Helix-D': Civil utility and rescue version, with upgraded avionics and search radar.

Ka-32T 'Helix-E': Civil utility version.

Ka-32IC: Civil Flying Crane.

Ka-32I: Civil utility version.

Ka-32AK: Fire-fighting version.

Ka-32AC: Civil version.

Status

In production.

Operators

India (Army), Russia (Army), Vietnam, Yugoslavia (Army)

Manufacturer

Kamov Aviacon (Kamovskiy Plant) in Kamov, Orel (Russia) design

Accommodation: two pilots, systems operator



Kamov Ka-28 'Helix'

(Tom Ripling)

Specifications (Ka-28)

Powerplant

Two Klimov TVO-117V turboshafts

Power: 4,180 shp (3,090 kW)

Dimensions

Length: 37 ft 1 in (11.3 m)

Rotor diameter: 52 ft 2 in (15.9 m)

Height: 17 ft 8 in (5.4 m)

Weights

Empty: 14,330 lb (6,500 kg)

Max T/O: 24,750 lb (11,000 kg)

Payload: 11,823 lb (5,000 kg)

Performance

Max speed: 168 mph (270 km/h)

Range: 432 nm (800 km)

Armament

Anti-submarine torpedoes, depth charges

Kamov Ka-29 (Russia) NATO reporting name 'Helix-B'

Type: Assault helicopter

Accommodation: Two pilots, 16 troops

Development/History

Capitalising on the success of the Ka-27 family, Kamov took the specialist assault helicopter version in the late 1980s. It was designed to operate off the Soviet Navy's amphibious landing ships, and is considered to be the 'twin' of the Mi-24, combining firepower with a troop-carrying capability.

Variants

Ka-29B 'Helix-B': basic assault transport version, also known as Ka-29UB.

Ka-29B(D): airborne early warning and surface surveillance version, redesignated Ka-31.

Ka-29K: anti-submarine warfare version based on Ka-29 airframe.

Status

In production

Operators

Russia (navy)

Manufacturer

Kamov Aviation (Sukhoi test and final) in Russia (Russia)

Design: Russia



Kamov Ka-29

(Rasovo/photographer)

Specifications

Powerplant

Two Klimov TVO-107V turboshafts

Power: 4000 shp (3026 kW)

Dimensions

Length: 32 ft 1 in (10.3 m)

Rotor diameter: 52 ft 2 in (15.9 m)

Height: 17 ft 8 in (5.4 m)

Weights

Empty: 12,120 lb (5500 kg)

Max takeoff: 37,775 lb (17,000 kg)

ayload: 16,535 lb (7500 kg)

Performance

Max speed: 174 mph (280 kmh)

Range: 249 mi (400 km)

Armament

Two 7.62 mm GSh-23ibre machine-guns in

deck gun pods, 9M114 'Sagger' (9K111

'Sagger') anti-tank and laser-guided anti-tank

missiles, free-fall bombs; 23-mm or 30-mm

gun pods

Kamov Ka-50/52 (Russia) NATO reporting name 'Hokum'

Type: Attack helicopter Accommodation: One pilot

Development/History

The Kamov OKB has had an interest in attack helicopters since the mid-1960s, when its design lost out to the Mil OKB's Mi-24 in the contest for the Soviet army's battlefield assault helicopter. Kamov resumed work in the 1980s, again with Mil as a rival, to fulfil a requirement for the Mi-24 replacement.

The Kamov Ka-50 first flew in 1982, and won the contest against the Mil's Mi-28 design due to its better agility, heavier armament and horsepower. However, the military establishment remained sceptical about the Ka-50's single-seat configuration, so work continued on the two-seat Mi-28. First unveiled in public in 1992, the Ka-50 is now being offered for export as the 'Werewolf' or 'Helicopter Soldier', although it has also been called the 'Black Shark' in promotional material.

The collapse of the Russian defence budget in recent years has left Russian army aviation in limbo, neither the Ka-50 nor Mi-28 have entered regular service, although 42 production versions of the Ka-50 have been completed and work continues on other versions, with a night-attack and two-seater version flying in prototype form.

The Ka-50 design is revolutionary, with the coaxial rotor allowing the traditional tail rotor to be dispensed with. By going for a single-seat design, Kamov OKB had to incorporate a significant number of automation devices, such as helmet-mounted sight, head-up display and computer recognition devices. Defensive equipment includes self-sealing fuel tanks and armoured cockpit. The pilot has an ejection seat, which



Kamov Ka-50/52 Werewolf

John Ripley

Specifications (for Ka 50)

Powerplant

Two Klimov TV3-117VK turboshafts

Power: 4380 shp (3266 kW)

Dimensions

Length: 52 ft 6 in (16.0 m)

Rotor diameter: 47 ft 2 in (14.5 m)

Height: 16 ft 2 in (4.9 m)

Weights

Empty: n/a

Max T/O: 23,670 lb (10,700 kg)

Warload: 6,614 lb (3000 kg)

Performance

Max speed: 163 mph (260 km/h)

Endurance: four hours with auxiliary tanks

Armament

One 30-mm 2A42 cannon; BM-30 Vinka-M (A-16) laser beam riding guided anti-tank missiles;

BM-114 Shrike (A-6 Sparrow) radar- and laser-guided anti-tank missiles; Kh-25MP (AS-12 Kegler) air-to-surface missiles; free-fight rocket pods, 23-mm and 30-mm gun pods; R-60M (A-8 Archer) or R-23 (A-11 Archer) heat-seeking anti-air guided missiles.



first triggers an explosive device to blow off the rotor blades prior to freeing the pilot safely away from the fuselage.

Variants

YAK: Initial prototype.

Ka-50 Werewolf (Black Shark/Helicopter Sash).

(nominally 'Vokum-A') V-005/006: Basic, single-seat version.

Ka-52 Alligator (nominally 'Vokum-B') V-006/07: Two-seat version.

Status

In limited production.

Kamov Ka-50/52 Werewolf
(Tim Ripley)

Operators

Russia (home).

Manufacturer

Progress Astrakhan Aviation Co (Russia) in Kamov OKB (Russia) design.

Mil Mi-2 (Russia/Poland) NATO reporting name 'Hoplite'

Type: Light helicopter

Accommodation: One or two pilots, eight passengers

Development/History

Under Warsaw Pact centralized defence plans, the Mi-2 Soviet plant was designated as the sole production site for the Mi-2 UEB Mi-2 design. The first Polish-built Mi-2 flew in 1965, and more than 5000 were built until production ceased in 1991. The light utility helicopter saw extensive service with Soviet and Warsaw Pact armed forces, including combat operations in Afghanistan and other trouble spots. Civil versions have been license produced in the USA.

Variants

Mi-2T: Unarmed utility/transport version.

Mi-2U: Dual control trainer.

Mi-2P: Agricultural crop sprayer.

Mi-2S: Medical evacuation version.

Mi-2TP: Armed version with 20 mm cannon pod and cabin machine gun.

Mi-2TMR: Soviet-made Armed reconnaissance version with 23 mm cannon and free-fight rocket pods.

Mi-2TMR: Armenian Armed reconnaissance version with Matigatu guided missiles.

Mi-2TMR: up-graded version with RARDT Sverd 2.

Mi-2C: Chemical: Chemical and nuclear survey and smoke layers.

Mi-2B: Upgraded version with improved electronics for export to Middle East.

Mi-2MK: naval version.

Mi-2RS: Reconnaissance version.

Mi-2RS: Chemical reconnaissance version.

Mi-2Sc: Dual control trainer.



(Tom Ryley)

Mil Mi-2 'Hoplite'

Specifications (for Mi-2T)

Powerplant

Two Lycoming TIO-150 (turbocharged).

Power: 100 shp (75 kW)

Payload: 1463 lb (660 kg)

Performance

Max speed: 124 mph (200 km/h)

Range: 257 nm (460 km)

Dimensions

Length: 37 ft 4 in (11.4 m)

Rotor diameter: 47 ft 6 in (14.5 m)

Height: 12 ft 3 in (3.7 m)

Armament

Free-fight rockets, gun and cannon pods; 9M14M Matigatu (9K-3 Sagger) wire-guided anti-tank missiles; 9M112 Sverd 2 (SA-7 Grail) anti-air missiles.

Weights

Empty: 3295 lb (1492 kg)

Max T/O: 8157 lb (3690 kg)



Mi-2/M: Survey version

Mi-2D: Ambulance command post.

Mi-2 Phazot Mine-laying version

Mi-2: Reconnaissance version.

Variant 51: East German surveillance version.

Variant 56: East German maritime version.

Variant 58: East German version.

Kamuflyotnyi: Version with Almico 250-C200 turboshaft, also known as Kamov Model 11.

Spartan Icarus: US-built version.

Status

Production suspended

Operators

Bulgaria (air force), Czech Republic, Estonia, Ghana, Greece, Republic, Iraq, Latvia, Libya, Lithuania, Nicaragua, Poland, Mongolia (air force), Romania (air force), Russia (military force), Slovakia, Syria (air force), Ukraine, USA (army)

Manufacturer

PZL-Swidnik (Poland) and Sparke Helicopter Company (USA) in a Mi-2 (Russia) design

Left

Mi-2 'Hoplite'
(Tom Roppon)

Right

Mi-2 'Hoplite'
(Tom Roppon)



Mil Mi-6 (Russia) NATO reporting name 'Hook'

Type: Heavy-lift helicopter Accommodation: Two pilots, flight engineer, navigator, radio operator, 65-75 troops, 41 stretchers

Development/History

Mi-6 giant heavy-lift helicopter made its first flight in 1957, and quickly set new standards in load carrying capacity. The largest helicopter of its generation, the Mi-6 saw widespread service with the Soviet army in Europe and Afghanistan.

Variants

Mi-6: Hook-A: basic version.

Mi-6P: Civilian passenger version.

Mi-6T: Military utility version.

Mi-6MP/MTP: 'Hook-B': Command/EW version.

Mi-6M/MAT: 'Hook-C': Command type also called Mi-32.

Mi-6P: Military rescue version.

Mi-6PZh/PZh: Fire fighting version.

Mi-6S: Medical evacuation version.

Mi-6Sp: Convertible version.

Mi-6T: Fuel transporter.

Status

No longer in production

Operators

Nigeria, Egypt, Ethiopia, Iraq, Laos, Peru (Peru Air Force), Poland (air force), Russia (army), Syria (air force), Vietnam.

Manufacturer

Revertent (Factory 160) (Russia) and Factory No 23 (Russia) to 54 (Russia) design



Mil Mi-6 'Hook'

(Foto: R. D. Smith)

Specifications (for Mi-6T)

Powerplant

Two Aviadvigatel/Polymer D-20V/MT turboshafts

Power: 10 250 shp (7620 kW)

Weights

Empty: 46 055 lb (21 240 kg)

Max: 103, 84 457 lb (46 400 kg)

Payload: 56 455 lb (25 600 kg)

Dimensions

Length: 101 ft 10 in (31.0 m)

Rotor diameter: 104 ft 10 in (32 m)

Height: 32 ft 4 in (9.86 m)

Performance

Max speed: 186 mph (300 km/h)

Range: 140 nm (100 km)

MIL Mi-8/17 (Russia) NATO reporting name 'Hip'

Type: Medium-lift helicopter

Accommodation: Two pilots, optional flight engineer, 24 troops, 12 stretchers

Development/History

The Mi-8 was the work horse of both the Soviet Union's armed forces and their Communist Bloc allies from the mid-1960s. Since the demise of the Soviet Union, the basic soundness of the design, and its low price, has enabled it to earn a major niche for itself in the world helicopter market. Although lacking the avionics of western machines, the glass-enclosed Mi-8 combines a useful carrying capacity with the performance to allow it to operate in the most extreme climatic regions.

The Mi-8 first flew in 1961, and has been continually upgraded throughout its long production life. The most significant improvement was the fitting of the up-rated Mi-8T/MT versions, which was designated Mi-17 for export customers - this version proved its worth in the 'Hot and High' conditions experienced during the 1979-89 Afghan war. The bloody conflicts in the fringes of the old Soviet empire and in the former Yugoslavia have seen the Mi-8 employed extensively in European war zones since 1991. The United Nations has also hired many Mi-8s to support its peace-keeping and humanitarian operations. To date, over 10,000 have been built for home and more than 60 export customers.

Variants

Mi-8 'Hip-A': Single engined prototype.

Mi-8 'Hip-B': Two-engined prototype powered by Klimov TV3 turboshafts.

Mi-8T 'Hip-C': Mass-produced version, powered by two Klimov engines, each rated to 1260 kW (1700 shp). Capable



MIL MI-8T 'Hip-C' on UN duty in Croatia

(Tim Ripley)

Specifications (for MI-8MT)

Powerplant

Two Klimov TV3-117 turboshafts

Power: 1046 shp (780 kW)

Performance

Max speed: 156 mph (250 km/h)

Range: 510 nm (920 km) with auxiliary tanks

Dimensions

Length: 50 ft 7 in (15.17 m)

Rotor diameter: 69 ft 10 in (21.3 m)

Height: 18 ft 6 in (5.65 m)

Armament

Door-mounted 12.7 mm machine gun, 9M117 Falanga (9K-2 Shatser) and 9M114 Malyutka (9K-3 Sagged) wire-guided anti-tank missiles.

9M114 Shatser V (9K-6 Sp-10) radio- and laser-guided anti-tank missiles; 9M113 Vinka (9K-12) laser beam riding guided anti-tank missile;

9K329 Igla V (SA-18 Griso) air-to-air missiles; three-flight rocket pods.

Weights

Empty: 14,200 lb (6,450 kg)

Max T/O: 28,455 lb (12,900 kg)

Payload: 13,200 lb (4,600 kg)

of being armed with free-fight rocket pods. Mi-8T uses Ukrainian engines.

Mi-8PM: Passenger and VIP transport version, also known as Mi-8P, 5 or P.

Mi-8TPS: Airborne liaison and command version.

Mi-8MV: Russian military designation for ex-export version with TV3-117M1 turboshafts. It has minor equipment changes. Ukrainian-built version known as Mi-8MV/V-171. Mi-8 MV/MV-1(-2)-3 are conversions to Mi-17 standard with port tail rotor.

Mi-8MTP: TV3-117MA powered version, with pressurized cabin.

Mi-8MV 'Hip-E': Armed version with 12.7 mm machine gun in nose and pilot-mounted Falanga missiles.

Mi-8MK: 'Hip-P': Armed export version with six launch rails for Matra-Sauveterre missiles.

Mi-8RL: Air accident investigation version.

Mi-8R/K: Reconnaissance/Artillery spotting version.

Mi-8MS: Search and rescue version.

Mi-8ZPU or VPC: Airborne radio or command post version.

Mi-8P: 'Hip-D': Airborne command post version.

Mi-8TS: 'Hip' in high desert version.

Mi-8MPP/VPU 'Hip-G': Airborne command post and radio relay version.

Mi-8MV 'Hip-J': Communications jammer (DURN) version.

Mi-8PTA 'Hip-K': Export electronic warfare version.

Mi-8PT: Polish airborne command post version.

Mi-8MA: Antisubmarine warfare version.

Mi-8MIL: Military ambulance version, also known as Mi-8M 'Helen'.



MIL MI-8TV 'Hip-H' of the Ukrainian Army Aviation on UN duty in Croatia (Tim Ripley)



Mil Mi-8M 'Hip-H' of Iraqi Air Force

(Tom Kipley)



Mi-8 Mi-17M 'Hip-M'

(Tim Ryley)



Mi-8 Mi-8T 'Hip-C' of Croatian Air Force seen over Bosnia

(Tim Ryley)

Mi-8TD: Liquid-cooled turboshaft version, with external tanks.
Mi-8-6AMKb: Night attack and combat rescue version with Shkval and Vinka guided missiles.

Mi-17 'Hip-H': Export designation for up-engined Mi-8MTV/MAK version with TV-117M turboshafts.

Mi-17P/M/PGP/H 'Hip-H/W': Export cargo-passenger version with large tailpods for external load on the rear of fuselage. Russian versions designated Mi-8MTS/MS/MSU/MSUW/MSUW/MSUW/MSUW/MSUW

Mi-17B-2: Czech electronic warfare version.

Mi-17M: Export version, with TV-117M engines, new clamshell rear cargo doors and loading ramp.

Mi-17R: Export version with new engines.

Mi-17-1M: High altitude operations version with TV-117M engines.

Mi-18: Proposed Korean-built Mi-17-1 version.

Mi-17-2M: Military transport and gunship version, with TV-117M engines.

Mi-17-TWA: Flying hospital version

Mi-172 (Mi-17M): Export version to Mi-8 MFI-3 standard.

Mi-17P: export passenger version

Mi-18: Re-engined derivative for original prototype, new cargo version

Mi-18t: Similar to Mi-9 carbamate-powered post

Status

In production.

Operators

Afghanistan, Algeria, Angola, Armenia, Azerbaijan,
Bangladesh, Belarus, Bosnia-Herzegovina, Bulgaria, China,
Croatia, Serbia [air force], Czech Republic, China, Colombia, Costa Rica,
Cuba, Czech Republic, Ecuador, Ethiopia [air force], Egypt,
Finland, Hungary, Iceland, Germany [air force], Indonesia,
India [air force], Indonesia [air force], Iraq, Kazakhstan, Laos,
Uganda [air force], Lithuania, Macedonia, Moldova,
Mongolia, Morocco [air force], Mexico [air force], Nicaragua, North
Korea, Pakistan [air force], Peru [air force], Poland [air force],
Sri Lanka, Serbia [air force], Romania [air force], Russia
[air force/missile force], Serbia [air force], Slovakia, Sri Lanka,
Sudan, Syria [air force], Tajikistan, Turkey [air force],
Ukraine, Ukraine [missile force], Venezuela, Vietnam,
Yemen, Yugoslavia [air force], Zambia, Georgia, USA [air force],
United Nations.

Manufacturer

Kazan Helicopter Plant [Russia/UKR], MI Moscow Helicopter
Plant [Russia], Progress Aviaexport Aviation Co [Russia], Ulan
Ude Aviation Plant [Russia], Dornier [Korea] to Olli Mil
[Russia] design.



MIL MI-8 AMTSh

(Tim Ripley)



MIL MI-8/17WD

(Tim Ripley)

Mil Mi-14 (Russia) NATO reporting name 'Haze'

Type: Land-based ASW helicopter

Accommodation: Two pilots, sonar operator, MAD operator

Development/History

The Mi-14 is an amphibious version of the Mi-8 developed for the Soviet Navy as a shore-based ASW and rescue helicopter. The first prototypes flew in 1972, and it has since been exported to a number of pro-Soviet states.

Variants

Mi-14: Prototype.

Mi-14PL 'Haze-A': ASW version with dipping sonar, search radar, retractable search radar and sonobuoy dispensers. The TV3-117V engine, rated to 1417 kW (1900 shp), was adopted during the later stages of production.

Mi-14PLM: Later version with better engines and systems.

Mi-14PS 'Haze-B': Mine-clearing version produced.

Mi-14PS 'Haze-C': Search and rescue version, with nose search light and anti-submarine gear removed.

Mi-14PK 'Haze-A': Pilot rescue training version.

Mi-14 'Illustrator': Mi-14 converted to fire bomber.

Status

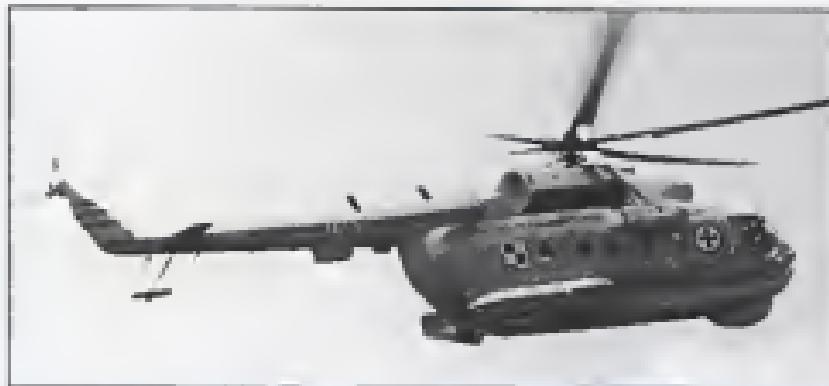
No longer in production.

Operators

Bulgaria [large], Cuba, Ethiopia, Libya [large], North Korea, Poland [large], Romania [large], Russia [large], Syria [large], USA [large], Yugoslavia [large].

Manufacturer

Kazan Helicopter Plant (Tatarstan) to Mi-8C (Russia)



Mil Mi-14PS 'Haze-C'

(Polish Model)

Specifications (for Mi-14PL)

Powerplant

Two Klimov TV3-117V turboshafts.

Power: 3400 shp (2556 kW)

Max takeoff weight: 14,000 kg (14,000 kg)

Payload: n/a

Performance

Max speed: 140 mph (225 km/h)

RANGE: 622 nm (1155 km)

Dimensions

Length: 60 ft 3 in (18.4 m)

Rotor diameter: 89 ft 10 in (27.3 m)

Height: 22 ft 9 in (6.9 m)

Weights

Empty: 25,000 lb (11,750 kg)

Armament

Anti-submarine: impediment, depth charges; door machine guns.

Mil Mi-24 (Russia) NATO reporting name 'Hind'

Type: Attack/assault helicopter

Accommodation: Pilot (rear), weapons operator (front), optional flight engineer, eight troops

Development/History

The distinctive Soviet assault helicopter was developed by Mil OKB in response to American experience in Vietnam. Some have called it a 'flying tank' because it has the best attack helicopter to feature heavy armor and be armed with a large caliber cannon. In Soviet/Russian service it is nicknamed the 'hunchback'.

The first prototype made its maiden flight in 1970 but the version boasted a full glass, or 'green house', cockpit, rather than the outer-tentacle tandem layout of later models. In 1974 the first production versions were spotted operating with Soviet troops in East Germany, and they were soon in widespread service throughout Eastern Europe.

The invasion of Afghanistan in 1979 gave the Mi-24 its first combat experience, and Soviet pilots soon came to value its heavy armored protection. Only the arrival of US-made Stinger missiles in the hands of Mujahideen rebels threatened Soviet air superiority, so a push programme to fit defensive systems to the Mi-24 was begun.

With the fall of the Soviet Union, the Mi-24 has seen extensive service as the way in the Caucasus - Russian Army Aviation used them to spearhead their invasion of Chechnya in 1994. Budget cuts mean Russian plans to replace the Mi-24 have yet to come to fruition, so it will have to soldier on for many years to come. In fact the Mi-24's appeal to export customers, western states and others have been integrated into the latest new-build versions.

Variants

Mi-24A/B 'Hind-B': Pre-production version, with TV-3-117



Mil Mi-24K 'Hind-G2' of the Ukrainian Army Aviation

(Tom Ryley)

Specifications (for Mi-24P)

Powerplant

Two Ivchenko TV3-117 series II turboshafts
Power: 4380 shp (3268 kW)

Performance

Max speed: 208 mph (331 kmh)
Range: 540 nm (620 km) with auxiliary tanks

Dimensions

Length: 51 ft 5 in (15.65 m)
Rotor diameter: 66 ft 9 in (20.3 m)
Height: 14 ft 1 in (4.3 m)

Armament

13.2 mm Gatling type gun or twin 23 mm cannons in nose; 9M17 Rokhaga (AT-2 Swatter) wire-guided anti-tank missile, 9M114 'Vigilant' (AT-6 Spear) radio- and laser-guided anti-tank missile; 9M136 'Valk' (AT-10) laser anti-tank guided missiles; 9M100 'Igla' (AT-10 Grapnel) and 9A-2300 air-to-air missiles; line-of-sight rocket pods; 25 mm or 12.7 mm gun pods; twin 30 mm dual-20-2 cannons; 30 mm grenade launchers; bombs, cluster weapons; mine dispensers.

Weights

Empty: 10,078 lb (4,570 kg)
Max WD: 26,455 lb (12,000 kg)
Max load: 5,290 lb (2,400 kg)

MIL MI-24 (Russia) NATO reporting name 'Hind'



engines, rated to 1200 shp.

Mi-24V 'Hind-K': Original production version with 'green house' front cabin, starboard tail rotor, TV-3-117 engines and Lukago turbines.

Mi-24U 'Hind-D': Unarmed training version of 'Hind-K'.

Mi-24U 'Hind-D': Early version to have tandem cockpit, 12.7 mm cannon and Falanga missiles.

Mi-24UB: Dual-control trainer with turret deleted.

Mi-25: Export version of Mi-24

Mi-24V 'Hind-E': Improved radio command-guided Urieme missiles. Powered by TV-3-117A engines. Known as Mi-24V in Polish version. Export version Mi-35.

Mi-24F 'Hind-F': Version of Mi-24D armed with high-precision twin 30 mm canons. Mi-35F export version.

Mi-24VP: Mi-24V with twin 23 mm canons in nose turret.

Mi-35M: Export version.

Mi-24B, BB, BCh (RCh) or BDR 'Hind-G': Chemical and nuclear surveying version.

Mi-24K 'Hind-G-2': Artillery fire correction version.

Mi-24VUB: Night attack version with western sensor, cameras and new Mi-26-style titanium rotor head.

Mi-24VUB: Unarmed export trainer.

Mi-24PS: Police/para-military version.

Mi-24L: Environmental research version.

Left: Mi-24V 'Hind-E'

(Tim Ripley)

Right: Mi-24V 'Hind-E'

(Tim Ripley)



MIL MI-24 (Russia) NATO reporting name: 'Hind'



Status

In production.

Operators

Afghanistan, Algeria, Angola, Armenia, Azerbaijan, Belarus, Bulgaria (air force), Cambodia, Croatia, Czech Republic, Ecuador, Ethiopia (South Africa), Hungary, Ireland, Hungary, India (air force), Iran, Kazakhstan, Laos, Libya (air force), Mongolia, Morocco, Nepal, Pakistan, Poland, Jordan, Russia (army), Rwanda, Sierra Leone, Slovakia, Sri Lanka, Sudan, Syria (air force), Thailand, Uzbekistan, Ukraine (army), Vietnam, Yemen, Georgia, USA (army).

Manufacturer

Reservei (Russia) and Progress Aviaexport Aviation Co (Russia) to Mil OKB (Russia) design.

Left: Mil Mi-24W 'Hind-E' of the Polish Air Force

(Tim Ripley)

Right: Mil Mi-28

(Tim Ripley)



Mil Mi-26 (Russia) NATO reporting name 'Halo'

Type: Heavy-lift helicopter

Accommodation: Two pilots, flight engineer, navigator, 80 troops, 60 stretchers

Development/History

Designed to replace the Mi-6, the Mi-26 is the most powerful helicopter in the world. It has a cargo carrying capacity equivalent to that of the C-130 transport aircraft. First flown in 1977, the Mi-26 entered Soviet Army Aviation service in 1985. The UN has chartered a number to support operations in Somalia and the former Yugoslavia.

Variants

Mi-26: Basic version.

Mi-26A: Civil version with D-136 engines.

Mi-26A2: Flying hospital version.

Mi-26M: Planned upgrade.

Mi-26T: Tanker.

Mi-26T: Upgraded version with D-136 engines.

Mi-26P: Proposed 20-seat passenger version.

Mi-26S: Export version.

Mi-26C: Upgraded navigation systems.

Mi-26TC: Hydro-ducted version with D-136 engines.

Status

In production.

Operators

India (Army), Peru, Russia (Army), Ukraine (Army), United Nations.

Manufacturer

Rosvertol (Russia) to Mil OKB (Russia) design.



Mil Mi-26 'Halo'

(Tom Ropley)

Specifications (for Mi-26)

Powerplant

Two DMS3 Progress D-136 free-turbine turboshaft.

Power: 22172 shp (16534 kW)

Weights

Empty: 62,170 lb (28,200 kg)

Max T/O: 123,450 lb (56,000 kg)

Payload: 44,680 lb (20,300 kg)

Dimensions

Length: 110 ft 8 in (33.7 m)

Rotor diameter: 105 ft (32 m)

Height: 26 ft 8 in (8.1 m)

Performance

Max speed: 183 mph (295 km/h)

Range: 403 mi (646 km)

Mil Mi-28 (Russia) NATO reporting name 'Havoc'

Type: Attack helicopter

Accommodation: Pilot (rear) and gunner (front)

Development/History

Superficially similar in appearance to the American Apache, the Mi-28 made its first flight in 1980. Since the aircraft lost the Soviet Army Aviation attack helicopter contract to the Ka-52, the Mi-28 has had a troubled history. The Russian Army Aviation has reportedly been persuaded to place an order for the aircraft, but funding difficulties have so far prevented series production taking place. The aircraft has been undergoing almost continuous development for over 16 years to allow it to fly armed attack missions at very low altitudes. Latest versions on display at western airshows include state-of-the-art night vision sensors and mast-mounted sights.

Variants

Mi-28: basic version.

Mi-28N: Night attack version with improved sensors and mast-mounted sight.

Status

In low rate production.

Operators

Russia (Army)

Manufacturers

Revertent (Russia) to a Mi OKB design.



Mil Mi-28N 'Havoc' with rotor mounted sight

(Tim Ripley)

Specifications (for Mi-28)

Powerplant

Two Klimov TVO-117V/M turboshafts

Power: 4,000 shp (3,000 kW)

Performance

Max speed: 165 mph (260 kmh)

Range: 240 mi (400 km)

Dimensions

Length: 55 ft 3 in (16.85 m)

Rotor diameter: 56 ft 5 in (17.2 m)

Height: 15 ft 9 in (4.87 m)

Armament

One 3A42 30 mm nose-mounted cannon; 9M113 Igla V (SA-16) missile and SA-2300 air-to-air missiles; 9M114 Shurm (FL-6 Sprut) anti-tank guided missiles; 9M1120 Vinka-M (AT-16) laser beam riding guided anti-tank missiles; free-fighters rockets.

Weights

Empty: 15,437 lb (7,000 kg)

Max T/O: 25,263 lb (11,500 kg)

Warload: 4,000 lb (1,814 kg)

Mil Mi-34 (Russia) NATO reporting name 'Hermit'

Type: Light utility helicopter

Accommodation: Two pilots, two passengers

Development/History

Designed as a light utility, observation, training and liaison helicopter for military, police, border guard and civil use, the Mi-34 made its maiden flight in 1985. It was the first Soviet helicopter to be capable of executing a loop or roll.

Production began in 1983, but funding problems slowed deliveries after six had been built. In 1997 production resumed after a corporate restructuring.

Variants

Mi-34: Basic version.

Mi-34V or V42: Twin-engined version, fitted with WZ-400 twin rotary engines, each rated to 160 kW (222 shp).

Status

In production.

Operators

Russia (Air Force/Army).

Manufacturer

Progress Antonov Aviation Co (Russia) and VIM Motor Car Works (Russia) to Mil DBS (Russia) design.



Mi-34

RU/TASSA

Specifications (for Mi-34)

Powerplant

VIM3 (Shvedenskiy) M-14V-26 air-cooled radial engine

Power: 320 shp (239 kW)

Weights

Empty: n/a

Max TOW: 2370 lb (1075 kg)

Dimensions

Length: 28 ft 7 in (8.71 m)

Rotor diameter: 32 ft 9 in (10 m)

Height: 10 ft 1 in (3.1 m)

Performance

Cruising speed: 112 mph (180 km/h)

Range: 234 nm (260 km)

Mil Mi-38 (Russia)

Type: Medium lift helicopter

Accommodation: Two pilots, 32 passengers

Development/History

Conceived as the replacement for the Mi-8/17 in the medium transport roles, the Mi-38 programme has not really got beyond the prototype stage because of lack of funding. Development began back in the mid-1980s, and a maiden flight was expected for 1993, but did not occur. It bears many similarities to the EH-101 Merlin.

The helicopter has many unique features, including a six-bladed main rotor, a delta 3 type tail similar to the Mi-26T, CRT cockpit displays and extensive use of composite materials. Cargo can be carried under-slung or positioned in the cabin via class-shelf rear doors and a loading ramp. Eurocopter is working jointly with Mil OKB and Kazan Helicopters on the programme.

Variants

N/A

Status

In pre-production

Operators

N/A

Manufacturer

Kazan Helicopter Plant (partnered) to Mil OKB (Russia)

Design



(Paul Jackson)

Model of the proposed Mi-38

Specifications (for Mi-38)

Powerplant

Two Klimov TV7-117V turboshafts

Power: 4636 shp (3456 kW)

Max 1/fb: 31 000 lb (14 000 kg)

Payload: 11 000 lb (5000 kg)

Dimensions

Length: 64 ft 2 in (19.50 m)

Rotor diameter: 64 ft 7 in (20.00 m)

Height: 16 ft 10 in (5.10 m)

Performance

Cruising speed: 155 mph (250 km/h)

Ranger: 200 nm (300 km)

Weights

Empty: n/a

Armament

N/A

Mil Mi-40 (Russia)

Type: Assault transport helicopter

Accommodations: Two pilots, 10 troops

Development/History

Intended as an assault transport version of the Mi-28 attack helicopter, it shares many of the systems of the Mi-28, including engine transmission, main and tail rotors.

Variants:

None

Status

In pre-production

Operators

None

Manufacturer

Assumed to be Mil OAO (Russia) design.



Model of the proposed Mi-40

(Paul Jackson)

Specifications (for Mi-40)

Powerplant

Two Klimov TV3-117MA turboshafts

Power: 4,000 shp (3,000 kW)

Dimensions

Length: 54 ft. 5 in (16.50 m)

Rotor diameter: 60 ft. 6 in (18.30 m)

Height: 14 ft. 6 in (4.40 m)

Weights

Empty: 16,100 lb (7,315 kg)

Max (OZ): 25,137 lb (11,400 kg)

Payload: 13,021 lb (5,900 kg)

Performance

Cruising speed: 113 mph (182 km/h)

Range: n/a

Armament

Anti-tank guided missile, fire-light rockets, gun pods.

Denel Aviation CSH-2 Rooivalk (South Africa)

Type: Attack helicopter

Accommodation: Pilot (rear), co-pilot/gunner (front)

Development/History

South Africa's Rooivalk (Red Kestrel) has its origins in an attack helicopter programme that commenced in 1981 in order to develop a successor to the Alouette III gunships then being used in Angola and South West Africa. The South African Air Force has ordered a squadron worth but defence cuts have put the order in doubt. Malaysia's new army aviation command may well be the first customer for the Rooivalk.

Variants

EDM: Experimental Development Model.

CSH-2: Basic production model.

ADM: Advanced development model.

Status

In pre-production

Operations (proposed)

Malaysia (lead), South Africa (in lead).

Manufacturer

Allo Aviation/Denel Aviation (South Africa)



(Denel Aviation)

Denel Aviation Rooivalk

Specifications (for CSH-2)

Powerplant

Two liquid-cooled

Power: 4000 shp (2983 kW)

Weight: 3622 kg (1321 kg)

Dimensions

Length: 24 ft 7 in (7.45 m)

Rotor diameter: 40 ft 5 in (12.00 m)

Height: 15 ft (4.6 m)

Max speed: 157 mph (250 km/h)

RANGE: 507 nm (900 km), 720 nm (1325 km)
with external fuel

Weights

Empty: 11 618 lb (5220 kg)

Max T/O: 20 723 lb (9300 kg)

Armament

One 20 mm GA-1 Rattler cannon; TOW 2A, TOW 2B or TOW 2B/AGM-114 laser-guided anti-tank missiles; VADS cluster or GBU air-to-surface missiles;
over-flight markers



Denel Aviation Oryx (South Africa)

Type: Transport helicopter

Accommodation: Two pilots, 20 passengers

Development/History

This South African-developed version of the Puma is being aggressively marketed by Denel to users needing helicopters optimised for fast and high-speed conditions. In many ways it is similar to the Super Puma because it uses the same powerplants, but Denel have gone further by modifying the tail section, plus building in the provision for an extensive array of armament. Previously known as Goucho.

Variants

Option 1: Gun-turret version

Option 2: Turret-mounted free-fight rocket launcher

Option 3: Nose-mounted free-fight rocket armament

Option 4: Anti-zebra quarry

Status

In production.

Operators

South Africa.

Manufacturer

Atlas Denel Aviation (South Africa)



Mock up of the stabilized sighting system fitted to an Oryx (AVL)

Specifications (for Oryx)

Powerplant

Two Turbomeca Makila 1A1 free turbines

Power: 3754 shp (2800 kW)

Max 0/0: n/a

Payload: n/a

Dimensions

Length: 50 ft 6 in (15.3 m)

Rotor diameter: 48 ft 2 1/2 in (14.6 m)

Height: 16 ft 10 1/2 in (5.14 m)

Performance

Cruising speed: n/a

Range: 300 nm (561.6 km)

Weights

Drywt: n/a

Armament

Free-fight rockets, 8 or 16 ZU-23-2 Swift or ZU-35 laser-guided anti-tank missiles; Barrels of Viper air-to-air missiles; 20 mm cannon-guns/turret

Westland Wasp (UK)

Type: Light general-purpose helicopter

Accommodation: One pilot, three passengers

Development/History

Once the primary day/night small helicopter of the British Royal Navy, the Wasp is now obsolete and is in the process of being phased out of service by its last remaining user.

Variants

Wasp HAS.1: Search and rescue.

Status

No longer in production.

Operators

Indonesia (navy), Malaysia (navy), New Zealand (air force)

Manufacturer

Swansea-Bre/Westland Helicopters (UK)



(Tim Ripley)

Westland Scout AH.Mk. 1

Specifications

Powerplant

One Rolls-Royce Bristol Mamba 503 turboshaft
Power: 710 shp (529 kW)

Max 1/2t: 5500 lb (2495 kg)

Payload: 1520 lb (688 kg)

Dimensions

Length: 30 ft 4 in (9.2 m)
Rotor diameter: 32 ft 3 in (9.8 m)
Height: 11 ft 8 in (3.5 m)

Max speed: 120 mph (193 km/h)

Range: 213 mi (340 km)

Weights

Empty: 3452 lb (1566 kg)

Performance

Max speed: 120 mph (193 km/h)

Range: 213 mi (340 km)

Armament

Max. 40 Hispano 45/2 mm-gatling machine
Max. 44 depth charges

Westland Lynx (Army version) (UK)

Type: Light multi-purpose military helicopter

Development/History

The British Army's primary light helicopter is another product of the Anglo-French Helicopter Agreement of 1967. Britain's Westland brought Lynx design to the table, and it duly became responsible for its development, production and marketing. Some 113 AH 1 were built for the British Army with dual landing gear, but export sales proved elusive. The British Army Air Corps and Royal Marines/Royal Navy later converted their fleets to armed helicopters (HEUARH) by fitting 25-mm/20-mm anti-tank missiles. A further 24 AH 2 light battlefield helicopter variants were produced from 1988 to equip 24 Armable Brigade.

Variants

AH 1: Original British Army utility version. Some examples armed with TOW missiles.

AH 10T: Interim armed version until AH 2 developed.

AH 2: Experimental version.

AH 3: Proposed Royal Marine version, not produced.

AH 3: British Army upgraded armed helicopter (AH/ARH) version with eight TOW missile tubes.

AH 4: British Army light battlefield helicopter version with Rolls-Royce Gnome 42-1 powerplant, each rated at 846 kW (1135 shp), tricycle under carriage and BERPW main blades.

Battlefield Lynx: Proposed export version with provision for Hellfire or HOT anti-tank missiles.

Battlefield 800: Proposed export version with UH/C 1600 engines.

Mk 24/25: Proposed Iraqi export version.

Mk 82: Proposed Egyptian export version.

Accommodation: Pilot, observer/gunner, 10 troops



Westland Lynx AH Mk 2

(Tim Rippy)

Specifications (for AH 1)

Powerplant

Two Rolls-Royce Gnome 32 turboshafts.

Power: 1600 shp (1343 kW)

Dimensions

Length: 40 ft 9 in (12.2 m)

Rotor diameter: 42 ft (12.8 m)

Height: 11 ft 6 in (3.5 m)

Weights

Empty: 6040 lb (2740 kg)

Max T/O: 10,600 lb (4800 kg)

Payload: 2000 lb (907 kg)

Performance

Cruising speed: 161 mph (258 kmh)

RANGE: 240 nm (393 km)

Armament

TOW and Improved TOW wire-guided anti-tank missiles, 12.7 mm or 20 mm short or pintle-mounted machine guns; laser-flare rockets.

Westland Lynx (Army version) (UK)

Mk 80: Proposed Saudi export version.

Mk 81: Proposed Qatari export version.

Mk 82: Proposed UAE export version.

Lynx AH1: Experimental advanced

composite helicopter with wings for additional lift.

Status

No longer in production.

Operators

16 (Army/Nav).

Manufacturer

Westland Helicopters (UK).



Westland Lynx AH Mk. 7
(Tim Ripley)

Westland Lynx (Navy version) (UK)

Type: Light multi-purpose naval helicopter

Accommodation: Pilot, observer/gunner, 10 troops

Development/History

Westland's development of the naval type has proved far more success than its effort with the army version. In addition to the 31 bought by the British Royal Navy, more than 200 have been sold for export, with new orders continuing to be secured.

Armed with the Sea Skua missile, the Lynx proved a potent ship killer both during the Falklands conflict and the 1982 Gulf War. After the Falklands, the Royal Navy began major upgrade programmes to improve the rotor blades, powerplant, sensors, weapon systems and electronic aids. This programme has continued through to the current HAS 8 standard, which is dubbed the Super Lynx.

Variants

HAS 2(H): French Navy anti-submarine warfare version, with OMLSS/Sigef OML 31W radar and Alouette chaffing system.

HAS 2: Original British Royal Navy version, with Ferranti Sea Sparrow radar, Boulton-Stephens sonar and Ferranti Instruments MAD.

HAS 3: Improved British version with two Rolls-Royce Gnome 41-1 875 kW (1,175 hp) engines.

HAS 3(H): Specialised British version for Arctic operations, from HMS *Endurance*.

HAS 3S: Specialised British version with surveillance and sensor communication equipment.

HAS 3HM: Improved British version for Gulf War with AGM-167 electronic counter-measures pod and anti-infrared jammers.

HAS 3CTC: Improved British version with central tactical



(GKN Westland)

Westland Lynx Mk 21

Specifications (for HAS 2)

Powerplant

Two Rolls-Royce Gnome 2 turboshafts

Power: 1800 shp (1342 kW)

Payload: 2000 lb (907 kg)

Dimensions

Length: 43 ft 9 in (13.3 m)

Rotor diameter: 42 ft (12.8 m)

Height: 11 ft 6 in (3.5 m)

Performance

Cruising speed: 161 mph (259 kmh)

Range: 240 nm (660 km)

Weights

Empty: 6740 lb (3060 kg)

Max TOW: 11 000 lb (4536 kg)

Armament

SL, 44, Mk 46 or Sting Ray anti-submarine torpedoes, Mk. 11 depth charges; Sea Skua anti-gated anti-ship missile, AGM-12 anti-gated missiles, 12.7 mm or 20 mm gun pods.

Westland Lynx (Navy version) (UK)

systems and flotation bag.

HAS.4 (HNC): Improved French Navy version with new GEM 41-1 engine, and gearbox.

Mk.21: Export version for Brazil, designated S-70-11.

Mk.21M: Export version of Super Lynx to Brazil.

Mk.22: Export version to Argentina. Often sold to Brazil and Denmark.

Mk.25/MH-14A: Export utility version for Netherlands.

Mk.27/Mh-14B: Export version for Netherlands with sensor.

Mk.28: Export version for Denmark.

Mk.29/MH-14C: Export version for Netherlands with MAD.

Mk.30: Export version for Norway.

Mk.37: Export version for Argentina.

Mk.68: Export version for Germany.

Mk.69: Export version for Nigeria.

Mk.90: Export version for Denmark.

HMS.8: Super Lynx upgraded version, with upgraded Rolls-Royce GEM 41-1 engines, HRRP rotor blades, Harpoon imaging sensor and improved electronic warfare systems.



Above:
Westland Lynx HAS
Mk.8/Super Lynx
(GKN Westland)

Left:
Westland Lynx
HAS/Mh.2 (FNT)
(Tim Ripley)

Mk. 95: Export Super Lynx for Portugal.

Mk. 99: Export Super Lynx for South Korea.

SH-14B: Export version for Netherlands with up-scaled Rolls-Royce Gnome RH-1 engines and full ASH for.

Super Lynx Series 200/300: Export version with TBB (TCT) 3000, improved sensors and 'glass' cockpit.

Status

In production.

Operators

Brand (navy), Denmark (navy), France (navy), Germany (navy), Malaysia (navy), Netherlands (navy), Nigeria (navy), Norway (navy), Pakistan (navy), Portugal (navy), South Korea (navy), UK (navy).

Manufacturer

Westland Helicopters/BAE Westland (UK).

Right:

Westland Lynx HAS Mk. 8/Super Lynx
(GKN Westland)



Kaman Seasprite (USA)

Type: Shipborne anti-submarine helicopter

Development/History

Making its first flight in 1977, the SH-2F version of the Sea Sprite utility helicopter was selected in 1980 by the US Navy for work on frigates, destroyers and cruisers in the anti-submarine role, under the LAMPS-I programme. It lost out to the SH-60 in the LAMPS-II contest, and the bulk of the US Navy's fleet have been either relegated to spare service or retired into storage. A programme to upgrade some surplus US versions to the anti-ship missile-armed SH-2G standard is underway, and the improved helicopter has recently found export success in Australia and New Zealand.

Variants

SH-2B: Shipborne utility helicopter for US Navy.

SH-2G: Initial winner of US Navy Light Airborne Multi-Purpose System (LAMPS) platform contest for embarked small ship helicopter. Powered by two T53-L-11T-1 powerplants. SH-2F: Improved version with 20% longer tail rotor blades, new search radar and towed MAD boom.

SH-2G Super Seasprite: Advanced version powered by two General Electric T700-GE-401 turboshafts, each rated to 1295 kW (1723 shp). It has improved mission sensors and weapon carriage capabilities.

SH-2G(A): Specialised anti-submarine warfare upgrade for Egypt.

SH-2G(M): Australian export version.

SH-2G(M): Proposed version for Malaysia.

Status

Work continues on SH-2G standard upgrades.

Accommodation: Two pilots, sonar operator, four passengers



Kaman SH-2F of HSL-34

(Jeremy Black/PA

Specifications (for SH-2G)

Powerplant

Two General Electric T700-GE-401 turboshafts.

Power: 3446 shp (3500 kW)

Dimensions

Length: 43 ft 6 in (13.24 m)

Rotor diameter: 44 ft 4 in (13.5 m)

Height: 15 ft 2 in (4.6 m)

Weights

Empty: 2200 lb (4172 kg)

Max T/O: 13,600 lb (6124 kg)

Payload: 4000 lb (1814 kg)

Performance

Max speed: 150 mph (244 km/h)

RANGE: 420 nm (385 km) with external tanks

Armament

Up to 50 torpedoes, depth charges, 176-mm floor guns, Penguin Mk 2 Mod 7 anti-ship guided anti-ship missiles, AGM-128/130/132. Maverick air-to-surface guided missile.

Operators

Argentina (Coast), Australia
Navy, Pakistan Navy, New
Zealand Air Force

Manufacturer

Kaman Aerospace (USA)



Right:

*Kaman SH-2F of HSL-34
Clement Fleet/AFP*

Bell Model 47 Sioux (USA)

Type: Light helicopter

Accommodation: Two pilots, one passenger

Development/History

One of the first helicopters to go into large-scale production after making its first flight in 1946, some 5,000 have since been built. Although it has now been withdrawn from regular service by most NATO airarms, it can still be found in use in obscure corners of Asia and South America.

Variants

H-13 Sioux: Basic US Army and USAF version.

TH-13/TH-1: TH-13/TH-1: US Navy trainer version.

HB-1/2: US Navy version for training and ice-breaking ship operations.

OH-13: Three-seat version.

UH-13: US Navy training version.

AB-47: Italian-built version.

AB-47G-2: UK-built version, developed Sioux AH-1Z.

Status

No longer in production.

Operators

Colombia, Congo (Shire), Greece [air force], Italy [army], Lebanon, Libya [air force], New Zealand, Pakistan [army], Paraguay, Peru [air force/navy], South Korea [army], Uruguay [navy], Zambia.

Manufacturer

Bell Aircraft Corporation/Bell Helicopter Company (USA), Agusta (Italy), Westland Helicopters (UK), Kawasaki Heavy Industries (Japan)



Bell 47G operated by the British Army as the AH.1 Sioux

Specifications (for Model 47G-3B-2A)

Powerplant

One Lycoming R0-405-FM piston engine

Power: 280 hp (209 kW)

Weights

Empty: 2033 lb (922 kg)

Max T/O: 2760 lb (1250 kg)

Dimensions

Length: 28 ft 7 in (8.6 m)

Rotor diameter: 39 ft 1 in (11.9 m)

Height: 9 ft 3 in (2.8 m)

Performance

Max speed: 105 mph (169 km/h)

Range: 215 nm (397 km)

Bell Model 204/UH-1 Iroquois (Huey) (USA)

Type: Light utility helicopter

Accommodation: Two pilots, seven passengers

Development/History

The first of the famous 'Huey' family of helicopters, which bear the brunt of the US Army campaign in Vietnam. Several thousand built for the US armed forces from 1956 through to the late 1960s.

Variants

UH-1A: Initial production version for US Army with Lycoming T53-L-1 turboshaft, rated at 815 shp, (605 shp). Capacity of six passengers. Source of 'Huey' nickname.

UH-1B: Enhanced version with capacity for seven passengers and revised main rotor blades.

UH-1A: Re-designation in 1962 of UH-1A.

UH-1B: Re-designation in 1962 of UH-1B.

UH-1C: Improved version of UH-1B, with T53-L-11 powerplant.

UH-1D: US Marine Corps version with hoist and twin .50 cal. machine guns.

UH-1E: US Marine Corps dual control trainer.

UH-1F: USAF ballistic missile surveillance version with General Electric T53-L-11, rated at 862 shp (620 shp).

UH-1G: Trainer version of UH-1H.

UH-1H: US Navy rescue version with hoist and T53-L-13 powerplant, rated at 864 shp (640 shp).

UH-1L: US Navy utility version with T53-L-13 powerplant.

UH-1M: US Navy training version with T53-L-13 powerplant.

UH-1N: US Army version with eight vision sensor (4 AB 204): Italian export version, with powerplant options including T53-L-13, rated at 962 shp (720 shp), Italian Lycoming T53-L-11A or Rolls-Royce Gnome II 1200, rated at



Aguila Bell AB 204B

(Courtesy Photo/AFM)

Specifications (UH-1C)

Powerplant

One Lycoming T53-L-11

Power: 1100 shp (820 kW)

Dimensions

Length: 43 ft 7 in (13.26 m)

Rotor diameter: 44 ft (13.41 m)

Height: 12 ft 7 1/2 in (3.84 m)

Weights

Empty: 6011 lb (2720 kg)

Max. (W): 10,200 lb (4627 kg)

Payload: 1361 lb (6180 kg)

Performance

Cruising speed: 140 mph (226 km/h)

Range: 333 nm (615 km)

Armament

One machine gun, machine gun pods; two light rocket pods; M6 44 (superior).



837 kW (1120 shp)

Hkp 30: Swedish designation of AB 204 AB 204AS: Italian-built naval version, with T53-G1-1 powerplant, rated at 863 kW (1150 shp)

Fuji-Bell 204B-2: Japanese-built version, also known as Hiyoko

Huey Iaq. UH-1C with up-rated engines

UH-1D: Research version.

Status

No longer in production

Operators

Austria, Colombia (air force), Honduras, Indonesia (army), Italy (army), Japan (army), Panama, Paraguay, Somalia, South Korea (army), Spain, Sweden (army), Thailand (army), Turkey (army) (navy), Vietnam.

Manufacturer

Bell Aircraft Company/Bell Helicopter Company (USA), Agusta (Italy), Fuji-Bell (Japan).

The Swedish army operates the AB 204 as the Hkp 30
Jeremy Black/APF

Bell Model 205/UH-1 Iroquois (Huey) (USA)

Type: Medium-lift helicopter

Accommodation: Two pilots, 12 passengers, six stretchers

Development/History

The first major upgrade of the ever popular 'Huey', which featured a stretched and enlarged cabin to boost carrying capacity. The first of 2500 ordered for the US armed forces entered service in 1963, whilst the last H-model was produced as recently as 1986. It went to continue in US military service until as late as the mid-1990s.

Variants

UH-1B: Original US Army version, with Lycoming T53-L-11 powerplant, rated to 820 kW (1100 shp). Capable of carrying 12-14 passengers.

UH-1C: Upgraded version for US Army, upgraded with T53-L-13 powerplant.

UH-1D: US Army medical and rescue version with hoist.

CH-11: Canadian training version, designated CH-113.

EH-1H: Electronic warfare 'Chuk-For' version.

HH-1H: USAF rescue version.

UH-1H Huey II: Commercial upgraded version with improved powerplant.

Huey 400: Commercial upgraded version with LHIC 1600 powerplant.

UH-1Z/UH-1Z00 Ultra Huey: Commercial upgraded version with General Electric T700-GE-701C powerplant, rated to 1400 kW (1900 shp).

UH-1H: Export-built version

AB 204A: Italian-built military version, designated EH-2, with T53-L-13 powerplant.

AB 204A-1: Improved Italian 204A.

AB 205B0: Prototype Italian version with two Geared H



Bell UH-1D of German Luftwaffe

(Tom Ryleigh)

Specifications (for UH-1H)

Powerplant

One Textron Lycoming T53-L-13 turboshaft

Power: 1400 shp (1944 kW)

Max T/O: 5600 lb (2540 kg)

Payload: 2800 lb (1270 kg)

Dimensions

Length: 41 ft 9 in (12.6 m)

Rotor diameter: 46 ft (14.0 m)

Height: 14 ft 6 in (4.4 m)

Performance

Max speed: 127 mph (204 km/h)

Rang: 276 nm (511 km)

Weights

Empty: 10,100 lb (4583 kg)

Armament

Two machine guns in door, optional search and machine gun pods.



1200 powerplant

All 205/M: Prototype turboshaft helicopter powerplants.

HE 108: Spanish designation for All 205.

Advanced 205D: Proposed Japanese upgrade

Stabes

No longer in production.

Operators

Argentina (air/regular force), Australia (army), Belarus,
Bangladesh, Bolivia, Brazil-Argentina, Brazil (air force), Brazil,
Canada, Chile (air regular force), Colombia (air force), Croatia,
Dominican Republic, Dubai, El Salvador, Germany (air/regular force),
Greece (air/regular force), Guatemala, Honduras, Indonesia (army), Iran
(army, navy, air force), Italy (army), Israel, Armenia, Japan (army),
Jordan, Mexico (air force), Morocco, Myanmar, New Zealand (air
force), Oman, Pakistan (army), Panama, Papua New Guinea, Peru (air
force, navy), Philippines, Saudi Arabia (air force), Singapore, South
Korea (army/air force), Spain (army), Suriname, Taiwan (regular
force), Thailand (air regular/air force), Tuvalu, Turkey
(air regular force), Uganda, UAE (Dubai), USA (air/regular force),
Uruguay (air force), Venezuela (regular force), Zambia, Zimbabwe.

Greece (air/regular force), Guatemala, Honduras, Indonesia (army), Iran
(army, navy, air force), Italy (army), Israel, Armenia, Japan (army),
Jordan, Mexico (air force), Morocco, Myanmar, New Zealand (air
force), Oman, Pakistan (army), Panama, Papua New Guinea, Peru (air
force, navy), Philippines, Saudi Arabia (air force), Singapore, South
Korea (army/air force), Spain (army), Suriname, Taiwan (regular
force), Thailand (air regular/air force), Tuvalu, Turkey
(air regular force), Uganda, UAE (Dubai), USA (air/regular force),
Uruguay (air force), Venezuela (regular force), Zambia, Zimbabwe.

Manufacturer

Bell Helicopter Company/Bell Helicopters Textron (USA), Agusta
(Italy), AIC (China), Bristow (Norway), Fuji-Helikopter

**Bell UH-1N of
US Army
Reserve**

Bell Model 212 UH-1N Iroquois (Twin Huey) (USA)

Type: Medium-lift helicopter

Accommodation: Two pilots, 14 passengers

Development/History

A twin-engined "Huey" was first proposed by Bell Helicopters, Pratt & Whitney Canada and the Canadian Government in 1968. The USAF took delivery of the first aircraft in 1970, and it soon became the standard utility helicopter of the US Marine Corps. Foreign sales followed in large numbers, with more than 800 being built to date.

Variants

UH-1N: Basic US Navy and Marine Corps version.

VH-1N: USAF and US Marine Corps VIP transport.

CH-135: Canadian version, later designated CH-135 Twin Huey.

Twin Huey-2000: Civil commercial version.

AB 212: Italian-built military version, with Pratt & Whitney Canada PW115D turboshaft-Pac powerplant.

AB 212ASW: Italian maritime version (dorsal fin fairing).

HU-1N: Spanish Army designation.

UH-1N (JHM) : Four-blade USMC upgraded version.

Status

In production.

Operators

Argentina (Argentine Air Force), Austria, Bahrain, Bangladesh, Bolivia, Brazil, Chile (air force), Dominican Republic, Ecuador (air force), El Salvador, Ghana, Greece (Hellenic Air Force), Guatemala, Guyana, Iran (Army/Army), Iraq, Israel, Italy (Armed Forces), Jamaica, Japan (Army), Lebanon, Malta, Mexico (air force), Morocco, Oman, Panama, Peru (air force),



Bell UH-1N of the USAF

Specifications (UH-1N)

Powerplant

Two Pratt & Whitney Canada PW115D Turbo

Turboshaft

Power: 1800 shp (1342 kW)

Weights

Empty: 6067 lb (2750 kg)

Max TOW: 11,760 lb (5290 kg)

Payload: 5300 lb (2360 kg)

Dimensions

Length: 43 ft 4 in (13.2 m)

Rotor diameter: 48 ft 2 in (14.7 m)

Height: 14 ft 10 in (4.53 m)

Performance

Max speed: 117 mph (188 km/h)

Range: 243 nm (450 km)

Bell Model 212 UH-1N Iroquois (Twin Huey) (USA)



Philippines, Saudi Arabia (air force), Singapore, Slovenia, South Korea (air force), Spain (military), Sri Lanka, Somalia, Sudan, Thailand (air force/aircraft), Tunisia, Turkey (army), Uganda, Uruguay (air force), Venezuela (army), Vietnam, Zambia, USA (DOD), UK (army), USA (military/aircraft), United Nations

Manufacturers

Bell Helicopter Company/Bell Helicopter Textron
USA/Canada, Agusta (Italy)

*Bell UH-1N of the USMC
(Tom Ropley)*

Bell Model 214 (USA)

Type: Medium utility and transport helicopter

Accommodation: two pilots, 16 passengers

Development/History

The first customer for the high specification version of the 'They' was the Imperial Iranian armed forces during the final years of the Shah's regime. Sales have followed to a number of customers who have been prepared to pay premium prices for a superior helicopter.

Variants

214A: Initial; Iranian-funded development, powered by two Lycoming T53-L1D, rated to 1520 shp (2050 kW).

214B: Bagh-e-Sabz (Cobra) version.

214C: Search and rescue version.

214ST: Low-winged version, powered by CT4-2As, with strengthened fuselage and composite rotor blades.

Status

No longer in production.

Operators

Bolivia, Colombia (air force), Ecuador, Iran (Army/Army Air force), Iraq, Oman, Peru (air force), Philippines, Thailand (Army), USAF (National), Venezuela

Manufacturer

Bell Helicopter Company/Bell Helicopters (Texas) (USA)



(Jeremy Black/APD)

Bell 214

Specifications (for 214ST)

Powerplant

two General Electric CT4-2A (turbo shafts)

Power: 1625 shp (121.2kW)

Weights

Empty: 9445 lb (4284 kg)

Max (O): 21811 lb (9845 kg)

Payload: 1700 lb (763 kg)

Dimensions

Length: 45 ft 2.5 in (13.72 m)

Rotor diameter: 52 ft (15.80 m)

Height: 15 ft 10.5 in (4.84 m)

Performance

Cruising speed: 161 mph (259 kmh)

Range: 463 nm (855 km)

Armament

Under-mounted machine guns

Bell Model 412 (USA)

Type: Medium utility and transport helicopter

Development/History

The most recent version of the 'Hercy' still manages to find customers around the world. A number of companies are also offering upgrade packages to older versions.

Variants

412: Basic production version.

412SP: Special Performance version, with improved fuel capacity, known as Aquila in Norwegian service.

412HP: Emergency medical services version, with improved transmission and PHG-100 Twin Pac.

Military 412: Armed version.

412EP: Enhanced performance version with additional fuel. Designated Chinook CH-1 in the Canadian service.

CH-146 Griffon: Canadian military version of 412SP.

Mk88-4 412: Italian-built version.

Mk412: Otokar: Italian-built military version. Designated TM-4 in Italian service.

Mk412 CH150: Italian-built ground surveillance radar platform.

Hkp 11: Swedish designation.

All 412 EP: Agusta-built version.

Status

In production.

Operators

Belarus, Bahamas, Canada, Colombia (air force), Guatemala, Guyana, Iceland (coast guard), Honduras, Indonesia (army), Italy (army/airforce force), Lesotho, Netherlands (air force),

Accommodation: two pilots, 14 passengers



Bell 412

Tom Ropley

Specifications (for 412HP)

Powerplant

One Pratt & Whitney Canada PT6A-13A Turbo

Twin Pac

Power: 1000 shp (734 kW)

Weights

Empty: 6654 lb (3018 kg)

Max T/O: 11 000 lb (5037 kg)

Performance

Cruising speed: 140 mph (225 km/h)

RANGE: 400 nm (744 km)

Dimensions

Length: 42 ft 4 in (12.87 m)

Width: 10 ft 10 in (3.30 m)

Height: 15 ft (4.57 m)

Armament

Door-mounted machine gun, various path rocket pods, Air-to-air and air-to-surface missiles

Norway, Peru (air force), Poland (air force), Saudi Arabia (air force), Slovenia, South Korea (air force), Sri Lanka, Sweden, Sweden (army), Thailand (air force/army), Uganda, UN (Dienst), United Nations, UN (Mali), Zimbabwe.

Manufacturer

Bell Helicopters Textron (USA/Canada), Agusta (Italy), EPH (Indonesia)



*Bell 412 of Dubai Police Air Wing
(Tim Ripley)*

Bell Model 206 JetRanger (USA)

Type: Light helicopter

Accommodation: Two pilots; three passengers

Development/History

The best-selling JetRanger first flew in 1966, and three years later the US Army began to take delivery of the OH-58 variant (see separate entry). It has since been adopted by a large number of armed forces around the world. Some 7,700 had been built by 1996.

Variants

Model 206A JetRanger: First production version, with Allison 250-C18 engine, rated to 235.5kW (317 shp).

Model 206B JetRanger II: Second production version, with Allison 250-C20, rated to 266 kW (400 shp).

Model 206B-3 JetRanger III: Improved version with 250-C20B powerplant.

Model 206Ae Chinook navy version, armed with torpedoes. 80-67 Czech: US Army version of JetRanger II, adopted for basic flight training under designation 11/20.

Model 206L-1 LongRanger: Stretched fuselage version of JetRanger II.

Model 206L-2 LongRanger II: Improved L-1, with Allison 250-C28B turboshaft, rated to 365 kW (490 shp).

Model 206L-3 LongRanger III: Improved version with Allison 250-C28B turboshaft rated to 405 kW (550 shp).

Model 206L-4 LongRanger IV: Canadian-built version.

Model 206L TwinRanger: Canadian-built twin-engined version.

Model 206L TrainRanger: Proposed military version of L-2.

Canberra CB-206J-III: Proposed gunship version for Iraq, built in Chile.

HH-57A Scalpanger: US Navy training version to 206A.



Bell 206 in United Nations service in Croatia

(Tom Sipley)

Specifications (206B-3 JetRanger III)

Powerplant

One Allison 250-C20B turboshaft

Power: 420 shp (300 kW)

Dimensions

Length: 21 ft 2 in (6.4 m)

Rotor diameter: 33 ft 4 in (10.2 m)

Height: 9 ft 6 in (2.9 m)

Weights

Empty: 1625 lb (737 kg)

Max T/O: 3700 lb (1675 kg)

Payload: Under-slung 1500 lb (680 kg)

Performance

Max speed: 140 mph (225 km/h)

Range: 365 nm (672 km)

Armament

One gun, 1000 rounds

**Standard**

TH-57B Seasenger: US Navy training version to 206B standard.

TH-57C Seasenger: US Navy training version to Jet Ranger II standard.

AB 206A-1: Italian-produced military version to 206A standard, designated AB-1 by Italian military.

AB 206A-2: Italian-purchased military version to 206B standard, designated AB-2 by Italian military.

AB 206C-1: Italian-modified A-1s upgraded to A-2 standard with -C70 engines.

ABp 6A: Swedish designation of Italian-produced 206A.

BB 17A: Spanish designation of AB 206A-1.

Zefar 206: Italian-purchased version of 206B-1.

Status

In production.

Bell 206L-4 LongRanger 4

(Bell Helicopters)

Operators

Austria, Bangladesh, Brazil (navy), Brazil, Cambodia, Chile (army/navy), Colombia (air force), Cyprus, Costa Rica, Ecuador (air force), Greece (army/navy), Guatemala, Guyana, Jamaica, Israel, Iran (army/navy), Italy (army), Libya (army), Malta, Mexico (air force), Morocco, Oman, Pakistan (army), Peru (army/navy/air force), Saudi Arabia (air force), Slovenia, South Korea (army), Sri Lanka, Sweden (army/navy), Tanzania, Taiwan (air force), Thailand (army), Turkey (army), Uganda, UAE (Dubai), USA (army/navy), United States, Venezuela (army/navy/air force), Yemen.

Manufacturer

Bell Helicopter Company/Bell
Helicopter Textron (USA/Costa Rica),
Austria (Italy), Custom Industries
(Oman)

Bell Model 206/OH-58 Kiowa (USA)

Type: Light observation and utility helicopter

Development/History

The US Army bought some 2000 versions of the OH-58 Kiowa from 1969 onwards to fly scout missions with specialized equipment fitted; the basic design has since undergone a number of upgrades to enhance its battlefield survivability.

Variants

OH-58A: Original US Army scout version.

OH-58B: Export version for Argentinian Army.

OH-58C: Upgraded US Army version with flat glass canopy and Allison 250-A120 turboshaft, rated to 213 kW (288 shp).

OH-58A: Canadian version to OH-58A standard, later redesignated OH-136 Kiowa.

Model 206B-1 Kiowa: Australian produced version, later renamed Kookaburra.

Status

No longer in production.

Operators

Austria, Australia (Army/Avon), Canada, USA (Army)

Manufacturer

Bell Helicopter Company/Bell Helicopter Textron (USA), Commonwealth Aircraft Company (Australia)

Accommodation: Pilot, co-pilot side-by-side, three passengers



US Army OH-58A Kiowa

(Jeremy Flock/APG)

Specifications (for OH-58A)

Powerplant

One Allison 250-A-120 turboshaft

Power: 217 shp (255.5 kW)

Weights

Empty: 1943 lb (878 kg)

Max T/O: 3000 lb (1361 kg)

Dimensions

Length: 32 ft 8.5 in (9.94 m)

Rotor diameter: 35 ft 4 in (10.77 m)

Height: 9 ft 6.5 in (2.91 m)

Performance

Cruising speed: 127 mph (196 km/h)

Range: 260 nm (481 km)

BELL Model 406/OH-58D Kiowa Warrior (USA)

Type: Light armed reconnaissance helicopter

Accommodation: Two pilots side-by-side

Development/History

The 'Kiowate' version of the OH-58, the Kiowa Warrior breathes an improved weapons and sensor fit to allow it to operate alongside the AH-64 Apache as part of joint air attack teams. The Army Kiowa Operational Improvement Program (KOIP) began in 1981, and the first helicopter entered service in 1985.

Variants

OH-58D Kiowa Warrior: US Army armed Scout version

Multi-Purpose Light Helicopter: US Army modification including folding rotor blades and tail to allow transport in C-130 transport aircraft

Proteo Chaser: Code-name for first a rebuilt fitted with Merlin and Super 90 rotors for slinging escort duties in Middle East in 1991

MH-60D/400C Combat Scout: Saudi Arab Forces version

Abu'l Qasas: prototype for QAH-20 anti-armor but no main-mounted sight.

OH-58Z: Stealth technology demonstrator.

Status

In production.

Operators

Saudi Arabia (Army), Thailand (Army), USA (Army)

Manufacturer

Bell Helicopter Textron (BHT)



Bell OH-58D Kiowa Warrior

(Bell Helicopter Textron)

Specifications

Powerplant

One Allison 250-C41 turboshaft

Power: 650 shp (485 kW)

Max (VR): 1200 lb (544 kg)

Warload: 2000 lb (907 kg)

Dimensions

Length: 34 ft 4 in (10.5 m)

Rotor diameter: 35 ft (10.7 m)

Height: 12 ft 10 in (3.9 m)

Performance

Max speed: 147 mph (237 km/h)

Ranger: 250 nm (463 km)

Weights

Empty: 3045 lb (1381 kg)

Armament

Stinger air-to-air missiles; AGM-114 Hellfire laser-guided anti-tank missiles; machine gun pods; free-fight rocket pods

Bell Model 209/AH-1F/G Huey Cobra (USA)

Type: Attack helicopter

Accommodation: Pilot, gunner in tandem cockpit

Development/History

Bell Helicopters first produced a gunship version of the Huey in 1963 as a private venture. Its distinctive tandem seating and nose turret have since been copied by attack helicopter designers around the world. Some 1000 G-models were bought by the US Army, and it proved very effective when used in action during the later years of the Vietnam war. The need to counter massed Soviet armored formations during the Cold War led to a series of upgrading programmes to provide the Cobra with the capability to fire 1080 war-guided anti-tank missiles. Service upgrades improved the performance at night and in bad weather. Israeli, Iranian and Kurdish forces have used 1080-armed Cobras in combat in the Middle East. US Army late-model Cobras were used in the 1991 Gulf War, and in conflicts in Somalia, Haiti and Bosnia.

Variants

Model 209: Original prototype.

AH-1G: Original US Army gunship version, with T53-L-13 turboshaft rated at 1044 kW (1400 shp).

TH-1G: Dual control trainer version.

AH-1H: Enhanced Cobras production version with 1080 missiles.

AH-1P: 1080 missile armed version.

AH-1Q: Upgraded version to allow 1080 missile carriage.

AH-1M: Upgraded version with T53-L-703 powerplant, improved AH-1S US Army version upgraded standard for OH-58D models, with T53-L-703 powerplant.

Production AH-1S: New-build version to AH-1G standard. Up-gunned version has 20 mm chaff/flare turret.



US Army AH-1G

Ulfhake Photo/AP

Specifications (for AH-1F)

Powerplant

One Pratt & Whitney T53-L-703 turboshaft.

Power: 1000 shp (1342 rhp)

Dimensions

Length: 12.11 m (40 ft)

Rotor diameter: 4.41 m (14.46 ft)

Height: 3.115 m (10 ft)

Weights

Empty: 4500 kg (9920 lb)

Max. Take-off weight: 10,000 kg (22,000 lb)

Performance

Cruising speed: 141 mph (227 km/h)

Range: 274 km (170 mi)

Armament

Four hard points; eight 1080 war-guided anti-tank missiles; two 8-shot rocket; M107 20 mm carriage on nose turret, 30 mm grenade launcher in nose turret.

AH-1F: Re-designation and upgrade of US Army Model 209, first seen flight cockpit glass, nose TOW sight and 152 L 702 powerplant. Current in-service version. Advanced AH-1 Model 209 King Cobra: Experimental version with single Lycoming T-53-L-1C powerplant.

Status

No longer in production

Operators

Bahrain, Brazil, Japan (Army), India, Pakistan (Army), South Korea (Army), Thailand (Army), Turkey (Army), United Nations.

Manufacturer

Bell Helicopter Company/Bell Helicopters Textron (USA), Fuji (Japan).

Right:

AH-1G Huey Cobra of the Maryland National Guard
Lenny Flack/AP



Bell Model 209/AH-1W Super Cobra (USA)

Type: Attack helicopter

Accommodation: Pilot, co-pilot/gunner in tandem

Development/History

US Marine Corps requirements for a twin-engined gunship to allow safe夜間 operation led to the halting of the AH-1W from 1971 onwards. Iran ordered an improved version but this was abandoned after the fall of the Shah in 1979. The US Marine Corps took over the programme which led to the 'Widow' version. It saw action during the 1991 Gulf War, claiming hundreds of Iraqi troops tanks with its laser-guided Hellfire missiles.

Variants

AH-1 Sea Cobra: US Marine Corps version with two Pratt & Whitney Canada T400-CP-400 turboshafts, rated at 1342 kW (1800 shp) each.

AH-1T International: Export version of AH-1W

AH-1T Improved Sea Cobra: Upgraded AH-1W for US Marines with improved T400-WW-407 powerplants, each rated at 1463 kW (1970 shp).

AH-1W Super Cobra: Later US Marine Corps version with improved T400-WW-401 powerplants, each rated at 1536 kW (1773 shp).

Cobra Vortex: Proposed UK version.

AH-1W (HGW): Proposed upgrade for US Marine Corps, providing laser main rotor blades and weapon system improvements.

AH-1W: Bonanza-produced version, with customised weapon system.

Model 209 King Cobra: Experimental upgrade with two projects and improved weapon system.

Model 249: Experimental four-blade version.



Bell AH-1W Cobra

(Bell Helicopter Textron)

Specifications (AH-1W)

Powerplant

two General Electric T400-GE-401 turboshafts
Power: 3646 shp (3320 kW)

Dimensions

Length: 46 ft 6 in (14.1 m)
Rotor diameter: 40 ft (12.6 m)
Height: 13 ft 6 in (4.1 m)

Weights

Empty: 10 260 lb (4627 kg)
Max TOW: 14 750 lb (6670 kg)

Workload: 4552 lb (2066 kg)

Performance

Max speed: 173 mph (278 km/h)
Range: 305 nm (567 km)

Armament

One three-barrel M137 20 mm gun in nose turret; four hard points; 10W wire-guided anti-tank missiles; Hellfire laser-guided anti-tank missiles; AIM-9L Sidewinder air-to-air missiles; gun pods; cluster bombs; free-fall rocket pods

Status

In production

Operators

USA (numerous), Thailand
(Army), Turkey (Army)

Manufacturer

Bell Helicopter Company (USA)
Helicopters SA (USA), IAR
SA Brasile (Brazil)



Bell AH-1W Cobra
(Bell Helicopter Textron)

Bell/Boeing V-22 Osprey (USA)

Type: Tilt-rotor transport

Accommodation: Two pilots, crew chief; 24 troops

Development/History

This revolutionary assault aircraft has gone through a prolonged development phase but has now progressed to production, with the first examples being delivered in 1995. The Osprey uses its rotors to take off vertically, and they then rotate to provide the power for horizontal flight. Current plans call for some 460 to be purchased by the US Marine Corps, in place of the CH-46 assault helicopters. The first unit, VMFA-224 'The White Knights', is scheduled to become operational at MCAS Cherry Point, North Carolina, by 2001. The USAF has a requirement for 50 Ospreys for special operations, estimated to be in service by 2006. The US Navy wants 48 Ospreys for combat search and rescue. First-rate initial production began in 1997 at the aircraft's aeronautical plant in Fort Worth, rising to eight in 2000, with a decrease on full production due that same year.

Variants

V-22A: Engineering and manufacturing development aircraft.

MV-22B: US Marine Corps assault production version.

MV-22C: Proposed initial US Navy anti-submarine warfare version.

CV-22B: USAF special operations production version.

HM-32B: US Navy combat search and rescue production version.

Bell/Boeing 608: Civilian passenger/MP transport tilt-rotor built to a smaller scale.

Status

In production



Bell Boeing V-22 Osprey

Bell Boeing

Specifications (V-22B)

Powerplant

Two Allison 412C-AD-100 turboshafts

Power: 12,300 shp (9123 kW)

Dimensions

Length: 77 ft 4 in (23.5 m)

Rotor diameter: 38 ft (11.6 m) each

Height: 17 ft 4 in (5.26 m)

Weights

Empty: 21,880 lb (14,463 kg)

Max: 38,100 lb (17,257 kg)

Payload: 20,000 lb (9072 kg)

Performance

Max speed: 185 mph (165 kmh) in helicopter mode; 316 mph (508 kmh) in fixed-wing mode

RANGE: 1200 nm (2224 km)

Armament

Door-mounted machine guns; maritime versions may be adapted to carry torpedoes and depth charges.

Operators

US Marine Corps

Manufacturer

Bell Helicopter Textron and
Boeing Helicopters (BHT)



Bell Boeing V-22 Osprey
(Bell Boeing)

Boeing CH-47 Chinook (USA)

Type: Heavy-lift helicopter

Accommodation: Two pilots, crew chief, 55 troops, 24 stretchers

Development/History

The 'mighty' Chinook first flew in 1962 to fulfil a US Army requirement for a heavy-lift helicopter. Viewed by the US Army as a 'flying truck', it proved its worth in Vietnam supporting an mobile bases and flying supplies and artillery pieces to remote jungle fire bases. The large under-slung load capacity of the Chinook soon led it to being nicknamed 'Hogtie' by US troops. Some 384 A-models were built for use during the Vietnam War, and more orders soon followed. A constant upgrade programme has significantly improved the capability of the US Army's Chinooks over the ensuing decades. Just under 600 were in service with the US Army, US Army Reserve and National Guard in 1991.

During the 1991 Gulf War CH-47Ds played a key role moving the six mobile bases of the 101st Airborne Division deep behind Iraqi lines. They also opened the way for US paratroopers from to enter Basra in 1998 by lifting position bridge sections into position across the Tigris River. Foreign customers have also found the Chinook much to their liking, and sales have been made both from the main plant in Philadelphia and other license production lines in Italy, Japan and the United Kingdom. In the Chinook, Britain using its aircraft extensively in the Falklands, Northern Ireland, the 1991 Gulf War and Bosnia. It can boast three evaluable during the 1990-91 war against Iraq, whilst Italy operated its helicopters, lastly in Somalia in 1992, and then during the evacuation of its citizens from Albania during the 1997 civil war.

Following Britain's example of using the Chinook to move



Boeing CH-47D

(Tim Ripley)

Specifications (for CH-47D)

Powerplant

Two Allison Lycoming T53-L-13 turboshafts
Power: 6000 shp (4474 kW)

Weights

Empty: 26 918 lb (12 210 kg)
Max TOW: 54 000 lb (24 494 kg)
Payload: 27 082 lb (12 284 kg)

Dimensions

Length: 51 ft (15.5 m)
Rotor diameter: 60 ft (18.3 m) each
Height: 19 ft 10 in (6.0 m)

Performance

Max speed: 177 mph (285 km/h)
Range: 613 nm (1135 km)

Armament

One machine gun

its air mobility brigade, the 82nd Airborne has reduced Chinooks to priority mobility for its new rapid reaction force. The US Army uses Chinooks for special forces operations, with night vision sensors and in-flight refuelling equipment fitted to allow low-level penetrations to land on any type of night. Britain's Royal Air Force is also procuring a version with similar capability for long-range medical search and rescue missions.

Boeing's Chinook won the battle for international orders against Sikorsky's Sea Stallion, with more than 1000 built, or ordered, for the US Army and export by 1997.

Variants

CH-47A: Original US Army version, with T55-L-1A powerplants, rated to 1000 kW (2200 shp).

CH-47B: Upgraded US Army version with T55-AC turbines, rated to 1102 kW (2850 shp) and increased rotor diameter.

CH-47C: Further improved US Army version with T55-L-1A turbines, rated to 1276 kW (2750 shp), and more fuel capacity.

CH-47D: US Army version with T55-L-212 turbines for better performance and triple+H hook for improved handling of aero-dense loads.

CH-47E: Canadian version to CH-47C standard.

CH-47F: Spanish version to CH-47C standard.

Chinook HC 1: British version to CH-47C standard but with triple+H hook capacity.

Chinook HC 1B: British version refitted with glass fibre blades.

Chinook HC 2: British version to CH-47D standard.



Boeing CH-47D

(Tim Ripley)

Boeing CH-47 Chinook (USA)

Chinook HC 3: British version to CH-47 standard
MH-47D Special Operations aircraft: Interim upgrade for US Army special operations units building of CH-47
MH-47E: Special forces version with in-flight refuelling, night flying capability and T700-GE-701 550 shp, each rated to 2064 kW (2750 shp).

Model 414: Export model to CH-47 standard

International Chinook: Export model to CH-47 standard
CH-47C: Phase II/III built version with T700-GE-701, powerplants and composite blades.

CH-47D: Argentine-built version to CH-47D standard
BH234MLB, Croatian version.

CH-47D: Improved Cargo helicopter upgrade for US Army, possibly to be designated CH-47F

Advanced Chinook: Proposed version with 2000 shp (2722 kW) class engines, redesigned rotors and additional fuel.

Status

In production

Operators

Argentina (as Fcoar), Australia (army), Egypt, Greece (army), Iran (army/air force), Italy (army), Japan (army/air force), Monaco, Netherlands, Singapore, South Korea (army), Spain (army), Taiwan, Thailand (army), UK (air force), USA (army)

Manufacturer

United Aircraft Corporation/Boeing Vertol/Boeing Helicopters (USA), Kawasaki Heavy Industries (Japan), Eads/AgustaWestland/Agusta (Italy).



Boeing CH-47 HC Mk 2

(Tim Rupley)



Boeing CH-47 HC Mk 2

(Tim Rixley)

Boeing 107/CH-46 Sea Knight (USA)

Type: Medium-lift helicopter

Accommodation: Two pilots, crew chief, 26 troops

Development/History

The canardized rotor-bladed Vertol Model 107 made its first flight in 1960 and entered service with the US Marine Corps in 1964. Nicknamed the 'Tug', it saw extensive service as an assault helicopter during the Vietnam War. Subsequent operations in Grenada, the Persian Gulf, Somalia, Liberia and Haiti have seen the CH-46 at the centre of the action. An upgrade programme kept the aircraft flying through the 1970s, 80s and 90s as the majority of the Marine Corps' embarked helicopter fleet. The Pentagon is keen to replace the ageing, and increasingly unreliable, CH-46 with the Osprey tilt-rotor. Delays in the V-22 programme mean the 'Tug' will have to soldier on into the 21st century.

US Navy fleet support squadrons are large users of the CH-46, operating from shore bases or supply ships. Foreign exports have been small, with Japanese production lines being the main centre of activity. One of the more famous exports of the aircraft was in use by the Swedish Navy to hunt Soviet submarines in the Baltic Sea during the 1980s.

Variants

107 Model II: Civilian version.

HH-1/CH-46A: Original US Marine Corps assault version with two T53-GE-10 turboshafts, each rated to 332 kN (750 shp).

HH-46B: US Navy utility and cargo transport version.

CH-46B: Updated US Marine Corps version with T53-GE-10 turboshafts.

CH-46D: Updated US Navy version with T53-GE-10 turboshafts.



Boeing CH-46E Sea Knight

(Tim Ripley)

Specifications (for CH-46E)

Powerplant

Two General Electric T53-GE-10 turboshafts.

Power: 332 shp (750 kW)

Max 107: 23 000 lb (10 410 kg)

Payload: 9 000 lb (4 082 kg)

Performance

Max speed: 153 mph (246 km/h)

Range: 206 nm (380 km)

Dimensions

Length: 44 ft 10 in (13.7 m)

Rotary diameter: 81 ft (24.6 m) each

Height: 16 ft 8 in (5.1 m)

Armaments

One machine gun

Weights

Empty: 13 000 lb (5 907 kg)



CH-46B: US Marine Corps rescue version.

CH-46F: Final production version for US Marine Corps, with improved avionics.

CH-46L: Upgrades B- and F-models for US Marine Corps, includes glass fiber rotor blades, improved avionics and R-R T53-L-1G powerplants.

CH-46F: VIP version for US Marine Corps.

CV-107 (J107): Japanese-built utility version, exported to Saudi Arabia.

Hkp 4: Swedish designation for CH-46.

CH-113 Labrador: Canadian search and rescue version.

CH-113A: Canadian army version.

Status

No longer in production.

Boeing CH-46D Sea Knight

(Tim Rixley)

Operators

Canada, Japan (Army/Naval/Sea Force), Sweden (Army), Saudi Arabia (Air Force), USA (Naval/Marine).

Manufacturer

Vertol Aircraft Corporation (Boeing Vertol/Boeing Helicopters) (USA), Kawasaki Heavy Industries (Japan).

Boeing/Sikorsky RAH-66 Comanche (USA)

Type: Reconnaissance/attack helicopter

Accommodation: two pilots in tandem

Development/History

The US Army's much troubled next helicopter replacement programme has incurred significant funding, but as yet production is still uncertain. Boeing and Sikorsky won the USA contract to replace the Chinook, OH-58 and OH-6 in 1991, their first prototype flying in 1996. They have been contracted to supply six aircraft for testing to the US Army by 2002 under a \$1.699 billion contract. The second aircraft is to fly in 1998.

The Comanche has a number of unique features, including a breakaway main rotor and pivoted tail rotor. It is the first helicopter to be developed using 'health' technology to minimise its radar cross-section, heat signature and engine noise.

Variants

All

Status

In pre-production

Operators

US Army

Manufacturer

Boeing Helicopters and Sikorsky Aircraft (USA)



Boeing/Sikorsky RAH-66 Comanche

(Boeing Sikorsky)

Specifications (for RAH-66)

Powerplant

Two T-700C 1400/HT-901 turboshafts

Power: 2000 shp (1493 kW)

Weights

Empty: 7740 lb (3515 kg)

Max: 10,112 lb (4587 kg)

Warload: 2017 lb (915 kg)

Dimensions

Length: 43 ft 4 in (13.2 m)

Rotor diameter: 39 ft (11.9 m)

Height: 11 ft 1 in (3.4 m)

Performance

Max speed: 204 mph (328 km/h)

Range: 1260 nm (2264 km) with external tanks

Armament

Under development



Boeing OH-6 Cayuse/MD500/MD530 (USA)

Type: Light utility helicopter

Accommodation: One or two pilots, four passengers

Development/History

The OH-6 Cayuse was developed by the Hughes Helicopter Inc for the US Army's Light Observation Helicopter (LOH) requirement in the early 1960s. Soon nicknamed the 'Loach', it saw a fine service in Vietnam in large numbers. Hughes, and later McDonnell Douglas, have continued to develop and upgrade the basic design, with over 4000 having been built by 1997.

Variants

Model 204/300C (Hughes): Two-seater of 500 series, which looks enclosed rear cockpit. Military versions designated OH-6A. Schenck Aircraft have since developed the design.

OH-6A (Hughes 300M) Cayuse: Original US Army light observation helicopter, known as the Loach.

OH-6B: He-engined version with T63-A-730 powerplant, rated to 31,322 lbW (420 shp).

OH-6C: Proposed five-bladed version with improved Allison 25-C20A engine, rated at 270 kW (360 shp). Commercial derivatives designated Model 300D and L.

OH-6D: Japanese-built version to OH-6A standard.

MH-6B: Special forces version.

MH-6C: Special forces version.

OH-6E: Special forces command post/electronic warfare version.

AH-6C: Special forces attack version.

Hughes 500: Civil version of the OH-6A/Model 300 with Allison 250-C18A turboshaft; rated to 206.5 kW (275 shp).

Model 500C: Export version modified for 'hot-and-high' operation.



Boeing MD500 in Israeli service

AM Spotters

Specifications (for Model 500E)

Powerplant

One Allison 250-C18B turboshaft

Power: 450 shp (335.6 kW)

Dimensions

Length: 23 ft (7.01 m)

Rotor diameter: 26 ft 5 in (8.05 m)

Height: 8 ft 9 in (2.67 m)

Weights

Empty: 1445 lb (655 kg)

Max T/O: 3000 lb (1361 kg)

Payload: 1714 lb (771 kg)

Performance

Max speed: 152 mph (242 km/h)

Range: 203 nm (403 km)

Armament

HW wire-guided anti-tank missiles; Slingshot anti-air missiles; 30 mm cannon pod; 7.62 mm machine gun pod; two light rocket pods; 40 mm grenade launcher; Mk 44 or 46 incendiary



Model 500M Defender: Commercial version of OH-6A.
OH-6D: Japan-built version based on up-engined Hughes OH-6, five-bladed main rotor and T-tail.

MD500M: Italian-built version based on up-engined Hughes OH-6.

Model 500MH/MSW: Export version for Spain with MAD boom.

Model 500MD Defender: Military version with armor and infrared exhaust suppression.

Model 500D Scout Defender: Armed reconnaissance version.

Model 500MD/MSW Defender: Maritime version with search radar and MAD boom.

Model 500MD/TCW Defender: Anti-tank missile armed version.

Model 500MD/MSW- THW Defender: Anti-tank missile version with mast-mounted sight.

Model 500MD Quiet Advanced Scout Defender: Four-bladed version with noise suppression.

Model 500MD Defender II: Armed version with quiet slow turning four-bladed rotor.

An OH-6 of the Danish army (APF)

Model 500E: Revised version with pointed nose, improved tricycle and Allison 25-C30B powerplant.

MD 500E: Italian-built version of 500E.

Model 500MG Defender: Specialist military version of Model 500.

Model 500UK Black Super: British-built military version.

MD500: Two-bladed main rotor fitted with pointed nose, powered by Allison 250-C30 turboshaft, rated to 317 kW (425 shp).

AH-6E: Special forces command post/doc/medevac version with Allison 250-C30 powerplant.

MH-6E: Special forces version with Allison 250-C30 powerplant.

AH-6P: Special forces attack version Allison 250-C30 powerplant.

MD500MG Defender: Military version with Allison 250-C30 powerplant.

MD500 Nightfire: Night attack version with improved sensors and powerplant.

MD500MG Paracutty Defender: Specialist version powerplant for police and border patrol.

MD500HT Lifter/MH-6H: Special forces version to MH-6H standard, with glass cockpit and "people plant".

AH-6U: Special forces attack version to MD500 standard.

MH-6U: Special forces version with improvements to MH-6E.

AH-6U: Special forces attack version to MH-6H standard.



MD530N in flight over Mesa, Arizona (APG)

Status

In production.

Operators

OH-6

Argentina (air force, navy), Bolivia (air force), Colombia (air force), Costa Rica, Croatia, Cyprus, Dominican Republic, El Salvador, Ireland, Greece (air force), Indonesia (air force), Israel, Italy (air force), Kenya, Mauritania, Mexico (air force), North Korea, South Korea (air force), Taiwan (air force)

MD500

Argentina (air force, navy), Bolivia (air force), Colombia (air force), Costa Rica, Croatia, Cyprus, Dominican Republic, El Salvador, Ireland, Greece (air force), Indonesia (air force), Israel, Italy (air force), Kenya, Mauritania, Mexico (air force), North Korea, South Korea (air force), Taiwan (air force)

MD530

Chile (air force), Colombia, Mexico (air force).

Manufacturers

Hughes Tool Company/Hughes

Helicopters Inc/McDonnell Douglas

Helicopters Corp/Hawker

Helicopters (Brazil)/Avibrá (Brazil/Argentina)
(Italy), Kawasaki Heavy Industries
(Japan), Korean Air (South Korea),
MDA (Argentina)



OH-6A Cayuse

(AP)

Boeing MD 520N/Explorer (USA)

Type: Light utility helicopter

Accommodation: One or two pilots, six passengers

Development/History

The NOTAR is a revolutionary tail-surfaceless helicopter concept, which has been under development since 1981. As yet it has not been officially adopted by a military user, although US Army special forces are understood to have used NOTAR versions.

Variants

OH-6A NOTAR: Experimental version, first ever NOTAR helicopter.

MD520N: Experimental version with NOTAR rotors/rotor, five-bladed main rotor and Allison 250-C28R-2 turboshaft, rated to 325.7 kW (435 shp).

MD Explorer: Twin-engined NOTAR version. Military version called Combat Explorer.

MD500N: Wide-body single-engined NOTAR version.

Previously designated MD500N.

MD500E: Eight-seat version of Explorer.

MD-520N-OF: Suggested US special forces NOTAR version.

Status

In production.

Operators

N/A

Manufacturer

Hughes Helicopters/McDonnell Douglas Helicopter Company/Boeing Helicopters (USA).

The revolutionary Boeing Combat Explorer is reported to be in service with the US Army Special Forces. (Boeing)

Specifications (for MD Explorer)

Powerplant

Two Pratt & Whitney Canada PW 206B3 turboshafts

Power: 1254 shp (908 kW)

Max T/O: 2073 lb (935 kg)

Payload: Under-slung 2000 lb (907 kg)

Performance

Max speed: 172 mph (276 km/h)

Range: 374 nm (692 km)

Dimensions

Length: 39 ft 4 in (11.98 m)

Rotor diameter: 33 ft 10 in (10.34 m)

Height: 12 ft (3.66 m)

Armament

MD-520 Helistar (two guided anti-tank missiles, machine gun pods; fire-light rocket pods)

Weights

Empty: 3315 lb (1455 kg)



Boeing AH-64 Apache (USA)

Type: Attack helicopter

Accommodation: Pilot (rear), co-pilot/gunner (front)

Development/History

After the successful combat debut of the Chinook in Vietnam, the US Army began formulating requirements in the early 1970s for advanced attack helicopters. Bell Helicopters and Hughes Helicopters Inc were selected to develop competing designs and the latter company was declared the winning contractor in 1978, although it was not until 1982 that the contract was signed for the first batch of heavily-armed and armored AH-64A Apaches. Hughes was bought by McDonnell Douglas in 1994, just as the first Apache was being delivered. Since then the US Army has received some 821 A-models, and more than 100 have been sold to export customers.

The AH-64 showed its potential during 1990 Persian Gulf operations during the late 1990s, but it was not until the 1995 US operation to seize Panama that the Apache first saw action.

In the 1991 Gulf War the Apache showed its full potential by flying deep strike missions behind Iraqi lines. A US Army task force used Apaches to fire the first missiles of Operation Desert Storm, destroying a key Iraqi radar site. Supporting the Coalition ground assault, Apaches helicopters accounted for more than 500 Iraqi tanks, 120 APCs, 30 air defense systems, 120 artillery pieces, 125 other vehicles, 10 radars, 50 bunkers, 10 helicopters and 10 aircraft on the ground. Eight AH-64s were hit by enemy fire, but only one was shot down, with its crew surviving. Israeli forces have used the Apache extensively against Islamic guerrillas in southern Lebanon, and on a number of occasions they have employed Helicopter gunships to 'targetedly' assassinate key enemy commanders.



Boeing AH-64A Apache of Royal Netherlands Air Force

(Boeing)

Specifications (for AH-64A)

Powerplant

Two General Electric T700-GE-701 turboshafts

Power: 3322 shp (2500 kw)

Dimensions

Length: 58 ft (18.5 m)

Rotor diameter: 48 ft (14.6 m)

Height: 12 ft 2 in (3.8 m)

Weights

Empty: 11,225 lb (5095 kg)

Max (R): 21,600 lb (9,750 kg)

Loaded: n/a

Performance

Max speed: 227 mph (365 km/h)

Range: 260 nm (483 km)

Armament

One 30-mm M230 Chain Gun; AGM-114 Hellfire laser and millimeter radar guided anti-tank missiles; Hellfire, Stinger or Matador/Heimdall air-to-air missiles; free-fall rockets.



Boeing AH-64A Apache
(Boeing)

The intimidating presence of low-flying Apache helicopters in Bosnia from 1995 onwards was considered by US Army commanders to be instrumental in the success of their peacekeeping mission.

The US Army is upgrading its Apache fleet by introducing the Longbow millimetric radar and new radio frequency guided version of the Hellfire missile, which effectively allows for very long range engagements to be fought at night and in bad weather. All the US Army fleet will be modified to allow use of the mast-mounted Longbow radar, but only some 237 radar sets are being purchased. The Netherlands and Britain are the first export customers for the Longbow Apache. To prepare for deployment of the highly capable AH-64D, the Dutch have already received a number of old US Army A models for use until new build machines are ready. Britain is setting up its own production line to produce the WAH-64Ds, which will feature unique engines, weapon systems and defensive aids - the first helicopter is due to make its private flight in March 1998.

Variants

WAH-64/McHughes Model 77: Experimental version.

AH-64A: Basic US Army version.

AH-64B/90: Proposed PAH version for German army.

AH-64B Longbow: Improved millimetric radar equipped version.

WAH-64H: US-built Longbow version with Rolls-Royce/Turbomeca RTM322 engines.

AH-64C: US Army version upgraded to allow installation of Longbow radar. Now to be designated D-models.



Boeing AH-64D Longbow Apache

©Boeing

Attack (Attack); back-up
Sea Apache; Proposed naval
version.

Status

In production

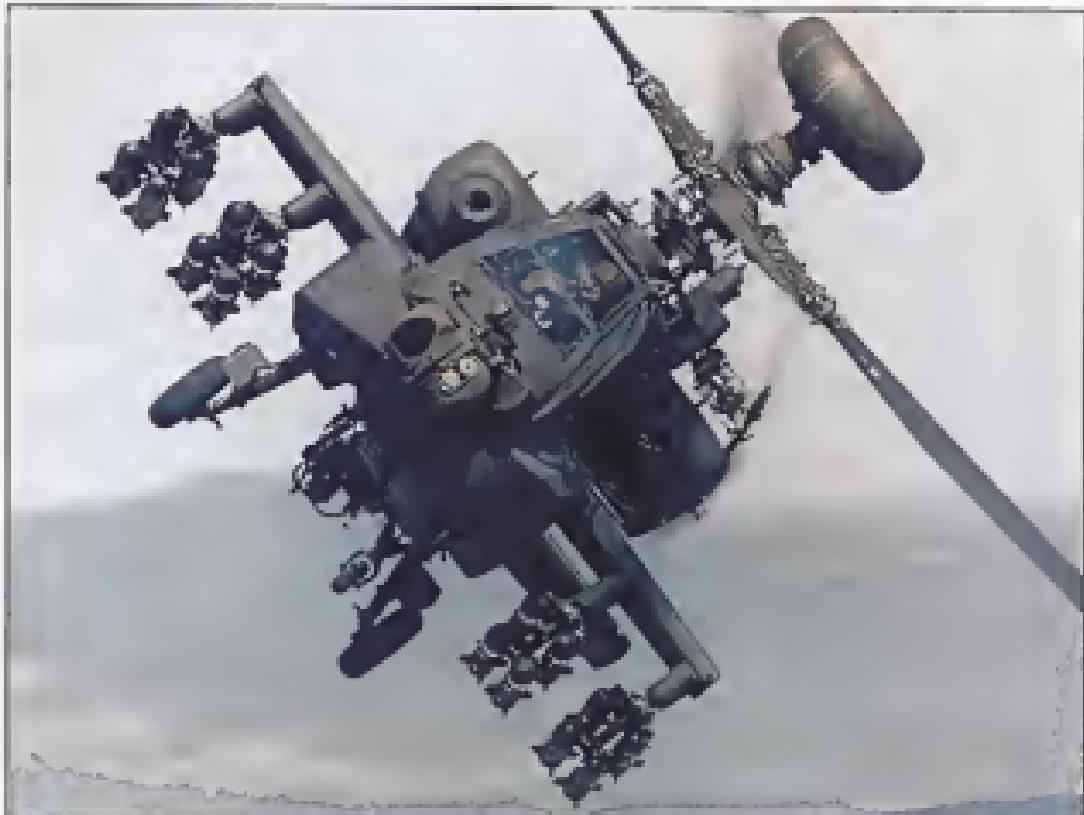
Operators

Egypt (as Army), Greece (Army),
Israel, Saudi Arabia (Army),
Milan (Italy) (as Army), USA
(Air Force), UK (Army), USA
(Army).

Manufacturers

Hughes Helicopters
McDonnell Douglas
Helicopter Company/Boeing
Helicopters (USA), Westland
Helicopter (UK).

Boeing AH-64D
Longbow Apache
(Boeing)



Sikorsky S-58 Choctaw/Wessex (USA)

Type: Medium-lift helicopter

Accommodation: Two pilots, optional crew chief, 16 troops

Development/History

The last version of the S-58 first flew in 1964, and the US Coast Guard operated large numbers until the CH-46 Heavy entered service in the 1980s. The British-built version, the Wessex, also saw extensive service. Westland improved the Sikorsky single piston-engined design by installing single- and then twin-turboshafts. Users are now withdrawing them from service, although Uruguay has recently bought up surplus British machines.

Variants (still in service)

Wessex HC.2: RAF utility and search version. Also operated by Uruguay.

Wessex HC.3: RAF transport and support helicopter.

Wessex HC.4: RAF Royal Flight VIP version.

Wessex SO: Rescue version used by Uruguay.

CH.3H: Hospital version.

CH.3D: Hospital version.

S-58: four-turboshaft engine-powered version.

Status

No longer in production

Operators

Argentina (Air Force), Chile (Air Force), Uruguay (Army), Iraq, Taiwan (Army), Thailand (Air Force), Turkey (Air Force).

Manufacturers

Yerkes Aircraft (PA), Westland Helicopters (UK)



Westland Wessex HC.Mk.5

(Tom Ropley)

Specifications (for Wessex HC.2)

Powerplant

Two Bristol Siddeley Orpheus Mk.10/11/12 turboshafts

Power: 2700 shp (2014 kW)

Weights

Empty: 13200 lb (5987 kg)

Max T/O: 19400 lb (8793 kg)

Payload: 6100 lb (2768 kg)

Dimensions

Length: 48 ft 4 in (14.7 m)

Length: 55 ft 10 in (17 m)

Rotor diameter: 67 ft (20.3 m)

Height: 16 ft 5 in (5.1 m)

Performance

Max speed: 140 mph (226 km/h)

Range: 214 mi (345 km)

Armament

7.62 mm gun

Sikorsky S-61/SH-3 Sea King (USA)

Type: Medium-lift/transport helicopter

Accommodation: Two pilots, (SH-3) two sensor operators, 26 troops

Development/History

The Sikorsky design made its first flight in 1958, and the American company made several licensees for the United States Navy during the 1960s. The SH-3 proved a very sound maritime helicopter, and NASA survey reduced it to large numbers from American and local production lines.

McDonnell Douglas in Britain began to develop its own versions from 1966, including anti-submarine, assault, airborne early warning and search and rescue. Production continued until the mid-1980s, with more than 300 being built for domestic and export markets.

Variants

YSS-2: Prototype version.

HSS-2/SH-3A: Original US Navy production version for anti-submarine warfare (ASW), powered by T-40-GE-8.

Initially rated at 9100 lb (4118 kg) GWR with dipping sonar and capable of carrying torpedoes or nuclear depth charges.

CH-3A/B: Utility version without ASW equipment for US Navy and USAF.

HH-3A: US Navy combat search and rescue version, featuring radio fuel tanks and Mokes equipment.

HH-3C: Experimental version, with jettisonable wings.

HH-3A: US Navy mine-clearing version.

V-3A: US Marine Corps version for Presidential transport.

SH-3H: Improved US Navy ASW version with T58-GE-10 engines and improved mission systems. License-built in UK, Italy and Japan.

SH-3D: US Marine Corps version for Presidential transport.



Sikorsky S-3G

(US Navy)

Specifications (for SH-3H Sea King)

Powerplant

Two General Electric T58-GE-10 turboshafts.

Power: 2000 shp (1500 kW)

Dimensions

Length: 54 ft 9 in (16.7 m)

Rotor diameter: 62 ft (18.9 m)

Height: 15 ft 6 in (4.7 m)

Weights

Empty: 11 065 lb (5020 kg)

Max TOW: 30 580 lb (14000 kg)

Payload: 8000 lb (3600 kg)

Performance

Max speed: 165 mph (265 km/h)

Range: 540 nm (1000 km)

Armament

Mk 44, 46, 48, A244B, Sonobuoy launcher, Mk 11 depth charges, Mk 57 and Mk 60 mine depth charges, Sea Eagle, AGM-114 Hellfire, Mine Mk 2 anti-ship missile, GAU-2 7.62 mm Minigun pods, machine guns.

with T-53-11-40 powerplant.

SH-3G: US Navy improvement of G-model with extra cargo and passenger carrying capacity.

SH-3H: US Navy improvement of G-model with improved mission system for ASW work.

UH-3H: US Navy utility version without ASW mission equipment.

SH-3D/T5: ASW version.

SH-3H ASW: Spanish navy airborne early warning version with Searchwater radar.

S-61MC: Export version for Denmark to SH-3H standard.

AS-61A-4: Search and rescue export version for Malaysia, known as Husk.

S-61B-3: Brazilian export version to SH-3D standard, later upgraded to SH-3H standard.

S-61D-4: Argentinean export version to SH-3D standard.

Italian-built versions

ASW-3D: Naval version, with Fiat-64-110 engines used to 1125 kW (1500 shp), ASW mission equipment and equipped to fire Exocet and Marte Mk 2 anti-ship missiles.

ASW-3H: ASW version with improved mission equipment.

AS-61-T5: VP transport version, designated AS-332/T5.

AS-61A-4: Export utility version with ASH-3D powerplant.

Canadian-built versions

CH-124CH-124H: ASW version to SH-3H standard.

CH-124B/C: Upgraded version with improved mission systems.



Sikorsky SH-3G

(US Navy)



Westland Sea King HC.Mk.4 'Jungle'

Royal Marines

Sikorsky S-61/SH-3 Sea King (USA)



Westland Sea King HC.Mk.4 'Junglet'

Japanese-built versions

S-61B: ASW version to SH-3A, later a S-61B-2 with improved mission systems was built to SH-3 standard.
S-61A/SH-3: Utility, Antarctic survey and rescue version.

British-built versions

Sea King HAS 1: ASW version with Rolls-Royce Gnome H1400 turboshaft is rated to 1050 kg (1400 shaft).
Sea King HAS 2: Improved ASW version with uprated Gnome H1400-15.
Sea King HC 4: Assault and long-haul transport version.
Sea King HAS 5: Improved ASW version with new radar and mission systems.
Sea King HAS 6: Improved ASW version.
Sea King HAR 2: Search and rescue version for RAF.
Sea King HAR 3A: Improved search and rescue version for RAF.
Sea King HAR 5: Royal Navy designation for its search and rescue version.
Sea King Mk 40: UK Ministry of Defence trials version.
Sea King Mk 41: Export version of Germany for search and rescue.
Sea King Mk 42: Export version for India to HAS 1 standard.
Sea King Mk 42A: Export version for India to HAS 2 standard.
Sea King Mk 42B: Export version for India with uprated Gnome H1400-17 powerplants.
Sea King Mk 42C: Export version for India to HAR 3 standard.
Sea King Mk 43/WB: Export version to Norway for search



Westland Sea King HC.Mk 8 'Jungle'

(film Noppey)

Sikorsky S-61/SH-3 Sea King (USA)

and variants

- Sea King Mk 45/W: Export version to Pakistan to HAS 12 standard
- Sea King Mk 47: Export AEW version for Egypt to HAS 2 standard
- Sea King Mk 48: Export search version for Belgium to HAS 3 standard
- Sea King Mk 50/W: Export version for Australia to HAS 2 standard
- Sea King AEW 2A: Anti-air early warning version with searchwater radar.
- Sea King AEW 2: Improved airborne early warning version with upgraded searchwater radar
- Commando Mk. 1 (Sea King Mk 70): Assault and troop transport version for Egypt
- Commando Mk. 2 (Sea King Mk 72): Assault and troop transport version for Egypt
- Commando Mk. 2A (Sea King Mk 92): Assault and troop transport version for Qatar
- Commando Mk. 2C (Sea King Mk 92): VIP version for Qatar
- Commando Mk. 21 (Sea King Mk 72): Electronic warfare version for Egypt
- Commando Mk. 3 (Sea King Mk 74): Naval version for Qatar, fitted to fire Exocet missiles.

Status

No longer in production

Operators

Argentina (navy), Australia (navy), Belgium, Brazil (navy),



Sea King HC.Mk 4 'Jungle' over Bosnia

IA (Phot) Terry Morgan

Canada, Denmark (air force), Egypt, Germany (navy), India (navy), Iraq, Iran, Italy (marines force), Japan (navy), Malaysia (air force), Norway, Pakistan (navy), Peru (navy), Qatar, Saudi Arabia (air force), Spain (navy), Thailand (navy), Venezuela (navy), UK (marines force), USA (navy).

Manufacturer

Sikorsky Aircraft (USA), Agusta (Italy), Westland Helicopters (UK), Mitsubishi Heavy Industries (Japan), United Aircraft (Canada).

Westland Sea King HC.Mk 4 'Jaggle' in service with the Royal Navy

(Media Production
CetLAND)



Sikorsky S-61N-1 Silver (USA)

Type: Passenger transport helicopter

Accommodation: Two pilots, 30 passengers

Development/History

A development of the Sea King largely for the civil market, this version has been employed by a number of military users, for troop transport and rescue work. Civil operators have also chartered them to military customers in the Middle East and the Falklands.

Variants

S-61L: Civil version

S-61NRL: Export search and rescue version for Argentina.

AS-61A-1: Italian-made export version for Malaysia.

Status

No longer in production.

Operators

Argentine Air Force, Malaysian Air Force, UK (Maritime), United Nations.

Manufacturer

Sikorsky Aircraft (USA), Agusta (Italy).



Sikorsky S-61N-1 Silver

Specifications (for S-61N)

Powerplant

Two General Electric T58-14B-1 turboshafts

Power: 3000 shp (2236 kW)

Dimensions

Length: 73 ft 10 in (22.3 m)

Rotor diameter: 62 ft (18.8 m)

Height: 17 ft 6 in (5.3 m)

Weights

Empty: 12,510 lb (5674 kg)

Max T/O: 22,800 lb (10360 kg)

Payload: 7,880 lb (3580 kg)

Performance

Max speed: 148 mph (235 km/h)

Range: 400 nm (740 km)

Sikorsky S-61/HH-3 (USA)

Type: Medium-lift transport helicopter

Accommodation: Two pilots, 30 troops, 15 stretchers

Development/History

Known as the Jolly Green Giant during the Vietnam War, the HH-3 revolutionized combat search and rescue work by being the first in-service helicopter to employ in-flight refueling. Eventually superseded by the bigger S-65 series in USAF service, the HH-3 found a niche in maritime rescue work with the US Coast Guard and Italian Air Force.

Variants

CH-3C: USAF utility and drone recovery version.

AS-61B: Pelican Italian-built search and rescue version.

HH-3E: Jolly Green Giant: USAF combat search and rescue version with in-flight refueling.

HH-3F: USAF special forces version with in-flight refueling.

HH-3G: Pelican: US Coast Guard search and rescue version.

HH-3L: USAF VIP transport version.

Status

No longer in production

Operators

Italian (air force), US (coast guard)

Manufacturer

Sikorsky Aircraft (USA), Agusta (Italy).



US Army HH-3F

(APU)

Specifications (CH-3E)

Powerplant

Two General Electric T53-L-11-5 turboshafts

Power: 3000 shp (2236 kW)

Dimensions

Length: 57 ft 3 in (17.4 m)

Rotor diameter: 62 ft (18.8 m)

Height: 16 ft 1 in (4.9 m)

Weights

Empty: 19,225 lb (8710 kg)

Max T/O: 37,050 lb (16,800 kg)

Payload: 5000 lb (2270 kg)

Performance

Max speed: 162 mph (261 km/h)

Range: 404 nm (748 km)

Armament

One machine gun

Sikorsky S-65A/CH-53 Sea Stallion (USA)

Type: Heavy-lift transport helicopter

Accommodation: Two pilots, crew chief, 37 troops, 24 stretchers

Development/History

Sikorsky's big litter first flew in 1964, and was quickly adopted by the US Marine Corps as its heavy assault transport. Some 124 D-models were bought by the Marine Corps, and later remained in service through to the 1990s. The USAF adopted the aircraft as its principal long-range special operations and combat search and rescue helicopter, integrating several upgrades to maintain its deep generation capabilities.

Variants

CH-53A: Original USMC version powered by General Electric T64-GE-10 turboshafts.

CH-53B: USAF training version similar in capability to CH-53A.

CH-53C: USAF combat search and rescue version with in-flight refuelling probes.

CH-53D: USAF rescue version with out in-flight refuelling probes.

CH-53E: Improved USMC version with upgraded T64-GE-410 engines, each rated at 2127 kW (2855 shp).

CH-53H: US Navy helicopter, powered by two T64-GE-410s each rated at 2026 kW (2700 shp).

CH-53J Pow Low III: USAF special operations version, fitted with in-flight refuelling, night vision equipment and terrain following radar and powered by two T64-GE-1A each rated to 2026 kW (2700 shp).

S-65C-20: Armenian export version built to CH-53C standard, later sold to Israel.

S-65C-3: Israeli export version similar to USAF CH-53Cs.



Sikorsky/TVW-Fokker CH-53G Sea Stallion serving with the United Nations Special Commission in Iraq after the Gulf War

(Tim Ripley)

Specifications (for CH-53A)

Powerplant

Two General Electric T64-GE-10 turboshafts.

Power: 1424 shp (6610 kW)

Dimensions

Length: 67 ft 9 in (20.67 m)

Rotor diameter: 72 ft 3 in (22.02 m)

Height: 24 ft 11 in (7.6 m)

Weights

Empty: n/a

Normal: 30,000 lb (13,607 kg)

Payload: External 13,000 lb (5,907 kg)

Performance

Max speed: 195 mph (314 km/h)

Range: 257 nm (463 km) with auxiliary tanks

Armament

1,07 mm or 12.7 mm chain gun



Sikorsky CH-53D Sea Stallion

(Tim Poglay)

Sikorsky S-65A/CH-53 Sea Stallion (USA)

CH-53 2000: Israeli upgrade also known as Yifter 2000; designed to extend life into the next century. Turkey is interested in buying this version.
CH-53: German-built version.

Status

No longer in production.

Operators

Germany (army, navy, air force), Israel, USA (air force/paramilitary).

Manufacturer

Sikorsky Aircraft (USA), MTM-Fokker (Germany).



**Sikorsky MH-53J
Pave Low**
(USAF/DoD)

Sikorsky S-80/CH-53E Super Stallion (USA)

Type: Heavy-lift transport helicopter

Accommodation: Two pilots, crew chief, 56 troops

Development/History

The S-80 Sikorsky Super Stallion will eventually replace all the most powerful heavy-lift helicopters in the world. The US Marine Corps and Navy began taking delivery in 1989, and some 127 were built until production ceased in 1996.

More than 100 aircraft used by the US Navy and Japanese Maritime Self-Defense Force are operated from amphibious warship ships or shore bases.

Variants:

CH-53E Sea Stallion: US Navy and Marine Corps Assault and heavy-lift version.

MH-53E Sea Dragon: US Navy mine-clearing version.

S-80E: Proposed export version of CH-53E.

S-80M-1: Japanese mine-clearing version.

Status:

No longer in production.

Operator:

USA (Marine Corps), Japan (Navy).

Manufacturer:

Sikorsky Aircraft (USA)



Sikorsky CH-53E Sea Stallion

(Tom Ripsley)

Specifications (for CH-53E)

Powerplant

Two General Electric T64-GE-416 turboshafts

Power: 13,140 shp (9700 kW)

Weights

Empty: 33,278 lb (15,072 kg)

Max (MTOW): 75,756 lb (34,340 kg)

Payload: (Internal/External) 36,488 lb (16,570 kg)

Dimensions

Length: 73 ft 4 in (22.3 m)

Rotor diameter: 75 ft (22.9 m)

Height: 20 ft 6 in (6.3 m)

Performance

Max speed: 195 mph (313 km/h)

Ferry Range: 1020 nm (1844 km)

Armament

2 TOW missiles, 12.7 mm don't guns

Sikorsky S-80/CH-53E Super Stallion (USA)



Sikorsky CH-53E Sea Stallion

Sikorsky S-80/CH-53E Super Stallion (USA)



Sikorsky S-80/CH-53E Super Stallion
(United Technologies/Sikorsky Aircraft)



Sikorsky S-80/CH-53E Super Stallion
(United Technologies/Sikorsky Aircraft)

Sikorsky S-70/UH-60 Blackhawk (USA)

Type: Medium-lift utility helicopter

Accommodation: Two pilots, crew chief, 14 troops

Development/History

In the early 1960s the US Army began looking for a UH-1-like utility replacement which would take into account many of the lessons learned from combat helicopter operations in Vietnam. Improved crashworthiness was a major criterion in the design, which first flew in 1974.

The first production aircraft flew in 1978, and soon the UH-60A was in widespread service with the US Army, seeing combat in Somalia in 1993. An improved version capable of lifting a HUMVEE or a 155 mm howitzer under slung was developed in the late 1980s, eventually being designated the UH-60L. In total the US Army has bought some 1400 aircraft, except plans for 2200. Low rate production continues for the US Army and export.

Variants

UH-60A: Original US Army utility version

UH-60B: SH-60S: Proposed general search/line radar version

UH-60E: US Army version with upgraded T700-GE-700C engines.

UH-60F: South Korean version to L-model standard

UH-60H: (Swath): Proposed medical evacuation version, with external load.

UH-60I: Quick Fix: Electronic warfare version

UH-60J: Quick Fix: Improved electronic warfare version

MH-60A: Velcro: US Army special forces version

MH-60L: Pow: Hawk: USAF special forces version with in-flight refuelling

UH-60M: USAF search and rescue version

MH-60R: US Army special forces version with in-flight



Sikorsky UH-60 Blackhawk

(Tom Murphy)

Specifications (for UH-60A)

Powerplant

Two General Electric T700-GE-700 turboshafts.

Power: 2344 shp (2420 kW)

Dimensions

Length: 50 ft 11 in (15.5 m)

Rotor diameter: 53 ft 8 in (16.4 m)

Height: 16 ft 10 in (5.1 m)

Weights

Empty: 16 284 lb (5000 kg)

Max: 30 250 lb (13,700 kg)

Payload: 9300 lb (4225 kg) unladen

Performance

Max speed: 164 mph (264 km/h)

Range: 315 nm (590 km); 1200 nm (2222 km) with maximum fuel

Armament

2 62-mm or 12.7-mm close guns and pods; two地狱火 (Hellfire) guided anti-tank missiles.



Sikorsky S-70 Armed Blackhawk

United Technologies/Sikorsky Aircraft

Sikorsky S-70/UH-60 Blackhawk (USA)



Sikorsky HH-60G Pave Hawk

(United Technologies Sikorsky Aircraft)

refuelling probe.

S-70A-11: US Army special forces version with in-flight

refuelling probe and uprated T700-GE-710C engines.

UH-60A: US Presidential transport version.

S-70A-1: Saudi land forces version.

S-70A-10: Saudi VIP transport/convoy version.

S-70A-11: Philippines export version.

S-70A-12: Australian-licensed version.

S-70A-13: Jordanian export version.

S-70A-12T: Japanese search and rescue version, designated UH-60J.

S-70A-14: Brazil export version.

S-70A-14C: Test bed for Rolls-Royce/Turbomeca RTM 322.

S-70A-17: Turkish export version.

S-70A-19C: Westland-produced version, designated WS-19.

S-70A-21: Egypt export version.

S-70A-24: Mexico export version.

S-70A-26: Monterrey export version.

S-70A-27: Hong Kong export version.

S-70C: Chinese export version.

S-70C-2: Rescue version with hoist used by France and Britain.

Status

In production

Operators

Australia (Army), Bahrain, Brazil (Army), Brazil, China, Colombia (Army/air force), Egypt, Israel, Korea (Army), Hong Kong, Japan (Army/air force), Jordan, Malaysia, Mexico,



Sikorsky UH-60 Blackhawk

(United Technologies/Sikorsky Aircraft)

Sikorsky S-70/UH-60 Blackhawk (USA)

Monaco, Philippines (air force), Saudi Arabia (air rep), South Korea (army), Vietnam (air force), Turkey (army), Thailand (army), USA (Army/Navy/US Air force)

Manufacturer

Sikorsky Aircraft (USA), Mitsubishi Heavy Industries (Japan), Westland Helicopters (UK), Hawker de Havilland (Australia), Korean Air (South Korea)



**Sikorsky S-70A
Blackhawk of Royal
Brunei Armed Forces**
(United Technologies/
Sikorsky Aircraft)

Sikorsky S-70B/SH-60 Seahawk (USA)

Type: Maritime helicopter

Accommodation: Two pilots, mission specialist

Development/History

Modified version of the S-70 series won the US Navy's LAMPS competition with a development contract being issued in 1987. The SH-60B has 80 per cent commonality with the SH-60, but includes many features necessary for operations at sea, including anti-corrosion treatment for the airframe, improved engines and a Bambi recovery device to lower the helicopter to a rolling ship deck in heavy seas. The US Navy has continued to develop the basic design, including a anti-submarine version with dipping sonar and a specialised combat search and rescue variant. Plans are now in hand to standardise the fleet under the SH-60B programme.

Status

In production.

Variants

SH-60B Seahawk: Original US Navy light multi-purpose system (LAMPS) SH-60B anti-submarine and anti-surface warfare helicopter, with APS-124 radar, MMW and sonobuoy launching systems.

SH-60F Ocean Hawk: Carrier-based (CV) version with new anti-submarine helicopter, with Bambi dipping sonar and provision for three Mk 50 torpedoes.

S-70B-1/SH-60B: Japanese-built version of SH-60B.

SH-60C: US Navy programme to standardise B, F and H versions.

S-60B-2 BAMS: Australian version with down-selectively-painted radar, sonobuoy and other systems. Also provision for Sea Skua and Penguin anti-ship missiles with ship missiles.



Sikorsky S-70B/SH-60 Seahawk of Greek Navy

(United Technologies/Sikorsky Aircraft)

Specifications (for SH-60B)

Powerplant

Two General Electric T700-GE-401C turboshafts

Power: 3800 shp (2834 kW)

Max T/O: 21 694 lb (9800 kg)

Payload: n/a

Performance

Max speed: 145 mph (234 km/h)

RANGE: 50 nm (37.5 km) for 3-hour loiter

Dimensions

Length: 50 ft 0.75 in (15.26 m)

Rotor diameter: 53 ft 8 in (16.31 m)

Height: 17 ft (5.18 m)

Weights

Empty: 13 645 lb (6170 kg)

Armament

162 mm and 12.7 mm close support ADM-110B

Penguin anti-ship missiles, Mk 46 or Mk 50 torpedoes, free-fight rockets.



S-70C(M)-1 Thunderhawk: Low-level version of SH-60F. Total conversion to Neysis intelligent role has taken place.
HH-60H Rescue Hawk: US Navy specialised combat search and rescue version, with extra armament and night vision system.
H6-60J Jayhawk: US Coast Guard search and rescue version.

S-70B-8: Greek export version.

S-70B-7: Italian export version with PT6A-30B engines.

CH-60: Proposed US Navy utility version for support and vertical replenishment.
Maple Hawk: Proposed Canadian rescue version.

Operators

Australia (navy), Greece (navy), Japan (navy), Spain (navy), United States (navy), USN (navy/civilian/guard)

Manufacturers

Sikorsky Aircraft (USA), Mitsubishi Heavy Industries (Japan), AS3A (Australia)

Sikorsky SH-60B Seahawk
(United Technologies/
Sikorsky Aircraft)



Sikorsky SH-60B Seahawk

(United Technologies/Sikorsky Aircraft)

Sikorsky S-76 (USA)

Type: Medium-lift utility helicopter

Accommodation: Two pilots, 14 passengers

Development/History

The private venture product has sold well to a number of civil and military customers around the world, but it has not found favour with the US armed forces.

Variants

S-76: Original version powered by Allison 250-C30 turboshafts, rated at 405 kW (540 shp).

S-76 Mk II: Improved version.

S-76 Utility: Basic version.

AH-64: Armed utility derivative, with provision for anti-armour, rockets and guns.

S-76JC: Version with 50% less fuel between armament powerplant.

S-76B: Production version with PW205-30A powerplant.

H-76B: Military version of S-76B, with weapons provision.

H-76C: Naval version.

ME-24: Spanish derivative.

Status

In production

Operators

Chile (Army), Costa Rica, Honduras, Hong Kong, Iraq, Japan, Jordan, Philippines (air force), Spain (air force), South Korea (army).

Manufacturer

Sikorsky Aircraft (USA) and Daejeon (Korea)



Sikorsky S-76C

(United Technologies/Sikorsky Aircraft)

Specifications (for H-76)

Powerplant

Two Pratt & Whitney Canada PT6A-13A turboshafts

Power: 1962 shp (1464 kW)

Max T/O: 11 700 lb (5267 kg)

Payload: n/a

Performance

Max speed: 170 mph (272 kmh)

Range: 317 nm (566 km)

Dimensions

Length: 44 ft (13.4 m)

Rotor diameter: 44 ft (13.4 m)

Height: 14 ft 5 in (4.4 m)

Weights

Empty: 6241 lb (2832 kg)

Glossary

AWW *Ambient air warning*

AWV *Air-to-vehicle vector*

ASW *Anti-surface warfare*

ASW *Anti-surface warfare*
avionics *Aeronautics electronics, such as
communications radio, radar,
navigation systems and computers.*

avionics *Radio system in aircraft
displaying heading and pitch change
more aircrafts are provided by the
flexibility of the structural material
and not by heavy. The rotor is rigid
carbonfiber fibre element of*

*carbonfiber fibre element as strength
element in composite.*

CAS *Clear air support*

CBU *Cluster bomb unit*

CFRP *Carbon-fiber-reinforced plastic.*

CH-46 *Countermeasures
command. Command from intelligence,
composite material. Made of two
components, such as fibers in
sheet which plus adhesive. Having
binding matrix*

detectors *Electronic high-lighting the picking
digital data between aircraft sensors
and system protection, usually MIL-
STD-1553B or ARINC 429 (one way)
and 649 (two way) sensors*

descent *Engines restricted to power less
than potential maximum (usually such
engine is that in cell)*

DF *Directional factor of direction finding
longitudinal helicopter blade with
many slender blades rotating in short
shaft*

DLE *Forward-looking infrared*

Dy-by-light *Flight control system in
which signals pass between computers
and actuators along fibre optic links.*

Dy-by-wire *Flight control system with
electrical coupling by without
mechanical connection between
control flying control and control
surface.*

g *Acceleration due to mass Earth
gravity due of a body in free fall is
acceleration due to rapid change of
direction of flight path*

DPS *Global Positioning System.
passenger Helicopter designed for
battlefield attack, normally with the
body carrying pilot and weapon
operator only*

hardpoint *Reinforced part of aircraft to
which external load can be attached,
e.g. weapons pylons.*

HMD *Helmet mounted display, hence
HMD = sight*

hot and high *Aerospace combination of
airfield height and high ambient
temperature, which brightness required
heat cell elements (HCE)*

by Hologram

HUD *Head-up display*

IR *Infrared sensor based on the*

IR *infrared.*

IRST *infrared search and track.
A-SATRS US Air Force/Army first*

*Surveillance Target Attack Radar
System in Boeing CH-47*

IRUDs *Joint tactical information
data system*

Keeler *Based fibre used as base of*

*high-strength composite material.
km/h kilometers per hour*

km/h *kilometers, the metric and for
measuring power output of jet engine
km/h = km per hour*

kw *Kilowatt, the metric and for
measuring power output of a
gas-turbine-driven engine*

km/hour of static thrust, the
measurement of a jet engine's static
thrust

LTV *low-light TV (the LTV, low-
light-level*

low observables *Materials and
structures designed to reflect aircraft
signature to all kinds*

m (metre), the metric unit of length

MAD *Magnetic anomaly detector*

MAD *Magnetic anomaly detector*

MADS *Multi-mounted sight*

MAD *Magnetic anomaly detector
Mach number*

mph (miles per hour)

MARL *Maximum take off weight
and Maximum altitude, 115000 miles (1.850
km)*

MES *Map-of-the-Earth (low-flying in
military aircraft using a bank of rows of
cells and rows etc)*

MES *Map-of-the-Earth*

opponents *Confined to all types and
directions of viewing and lighting
systems*

port left side, looking forward
pylon *Structure (wing) used to
external load (propeller nacelle, drop
tank, bomb etc)*

radii. The distance an aircraft can fly
from base and return without
intermediate landing

RAD *Radar absorbent material.*

radar *see heterodyne radio*

RHV *Remotely-guided vehicle*

RHV *Reach and rescue*

RJ *synthetic aperture radio,
sig. Shaft frequency, measure of power
transmitted via rotating shaft
signal. Signals with phase.*

signature *Characteristic "fingerprint" of
all electromagnetic radiation (radio, IR
etc)*

single-shaft *Gas turbine in which all
compressors and turbines are on
common shaft rotating together.*

SQR *Self-seeking reference radio*

stabilizer *See Diver, horizontal stabilizer
= stabilizer*

stall *Flight rate, looking forward
1 hour, 1 kiloparsec, 1000 kg
lift-rate aircraft with fixed wing and
wings that lift up the aircraft and
speed up the flight.*

SO *take-off*

so *Impulse (long) ton = 1.000 l or
2240 lb, US (short) ton = 0.987 l or
2000 lb*

tail-shaft *Gas turbine in which as
much energy as possible is taken from
gas jet and used to drive helicopter
rotor*

TAU *Measured on vehicle*

target *Small missile arrested, usually
sharply rebounded and often
overturns, or tip of wing.*

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